

# IPHA TECHNICAL SEMINAR 2017

October 25–26. Tallinn, Estonia

## Flexural strength and camber

**Kim S. Elliott**

Consultant | UK

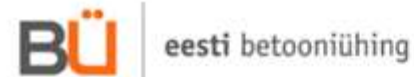


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PRECAST SOFTWARE  
engineering

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# Prestressed Concrete Hollow Core Units

## Flexural strength and deflections

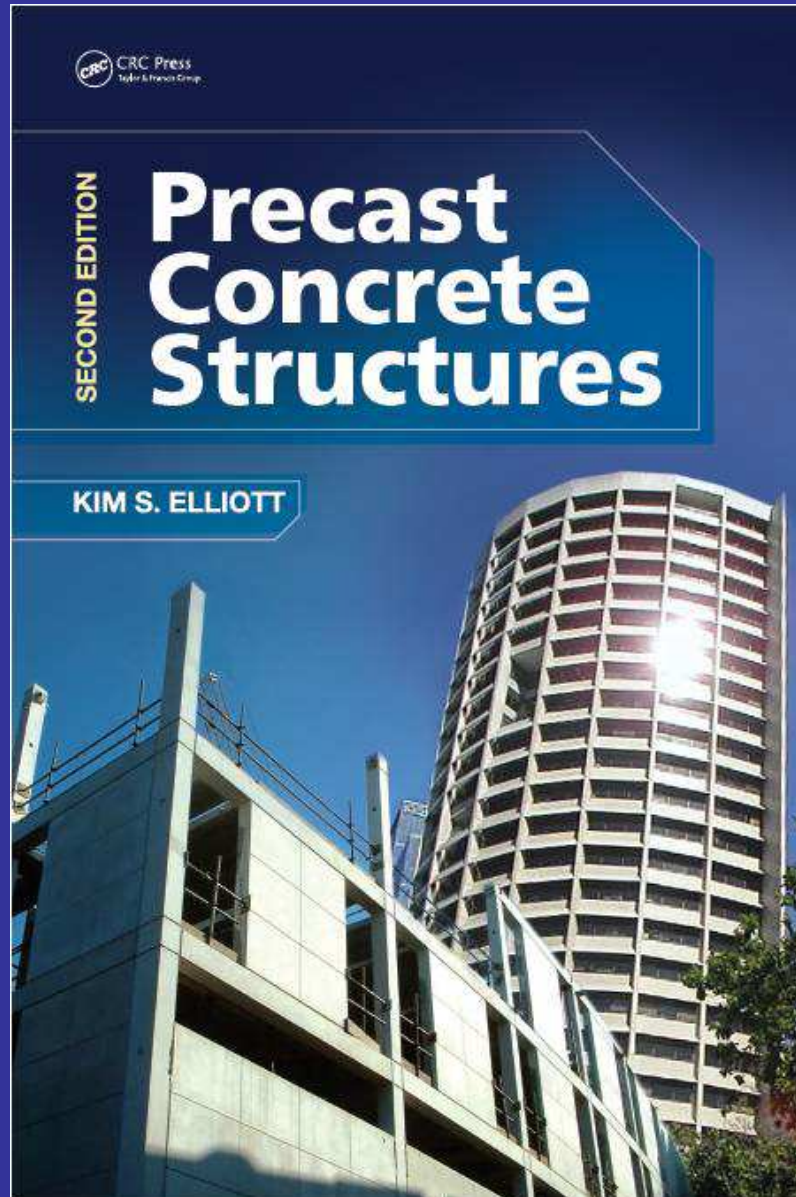
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**Dr Kim S Elliott, UK**

**IPHA Technical Seminar – Tallinn – 25-26 October 2017**

# Precast Concrete Structures 2<sup>nd</sup> ed.



700 pages with about 200 pages on precast floors

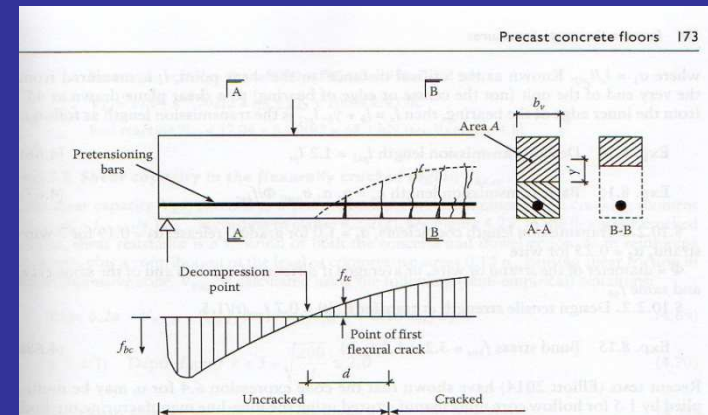


Figure 4.25 Principles of shear resistance for prestressed concrete elements.

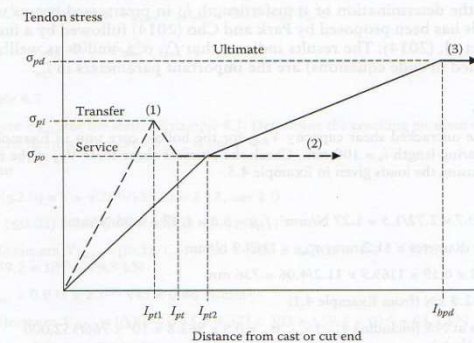


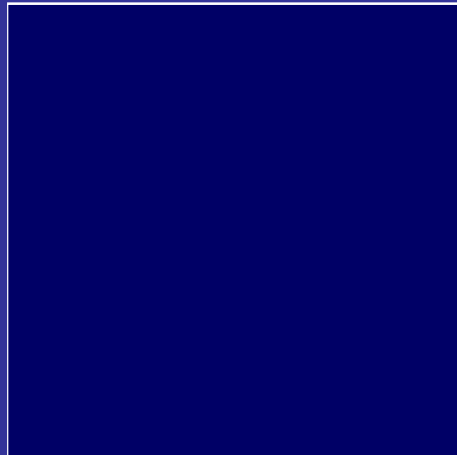
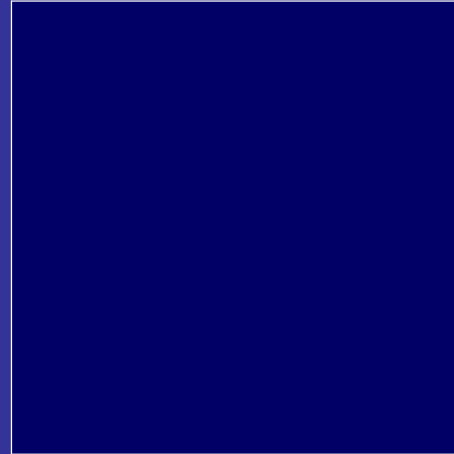
Figure 4.26 Development of stresses in the transmission zone  $l_{pt2}$  and anchorage zone  $l_{bpd}$  of prestressed members at (1) release of tendons, (2) after all losses and (3) ultimate. (Based on BS EN 1992-1-1, Design of concrete structures – Part 1-1: General rules and rules for buildings, BSI, London, Figs. 8.16 and 8.17.)

Design rules recognise the fact that the critical shear plane may occur in the prestress development zone where  $\sigma_{sp}$  is not fully developed. It is known that prestressing forces develop somewhere between linearly and parabolically, although BS EN 1992-1-1, Figs. 8.16 and 8.17 (combined here in Figure 4.26), adopts a linear development of stress in service and bi-linear at ultimate. Therefore, a reduced value  $\alpha_1 \sigma_{sp}$  is used up to  $\sigma_{po}$ ,

# Syllabus

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**Definitions.  
Introduction.  
Concrete  
and strands.  
Cover.**

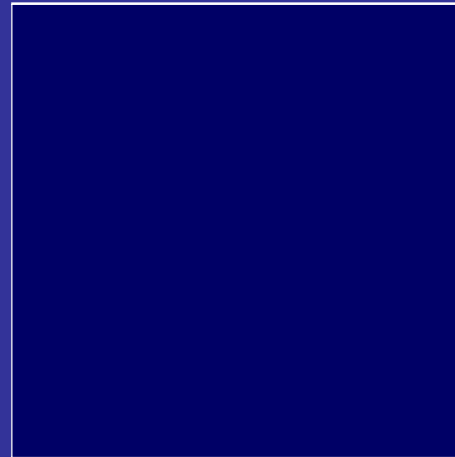
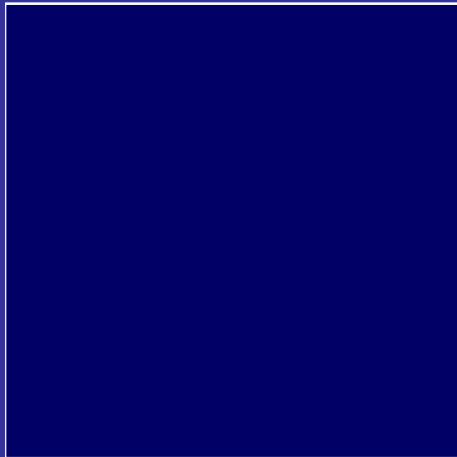


# Syllabus

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**Definitions.  
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**Prestress.  
Losses.  
Limit  $f_{ctm}$ .  
Moment of  
resistance.**

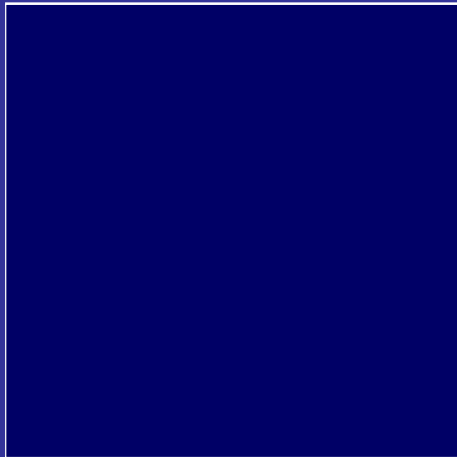


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**Ultimate  
strength.  
Equilibrium.  
Compatibility.  
 $M_{Rd}$**

# Syllabus

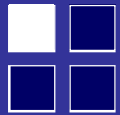
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**Definitions.  
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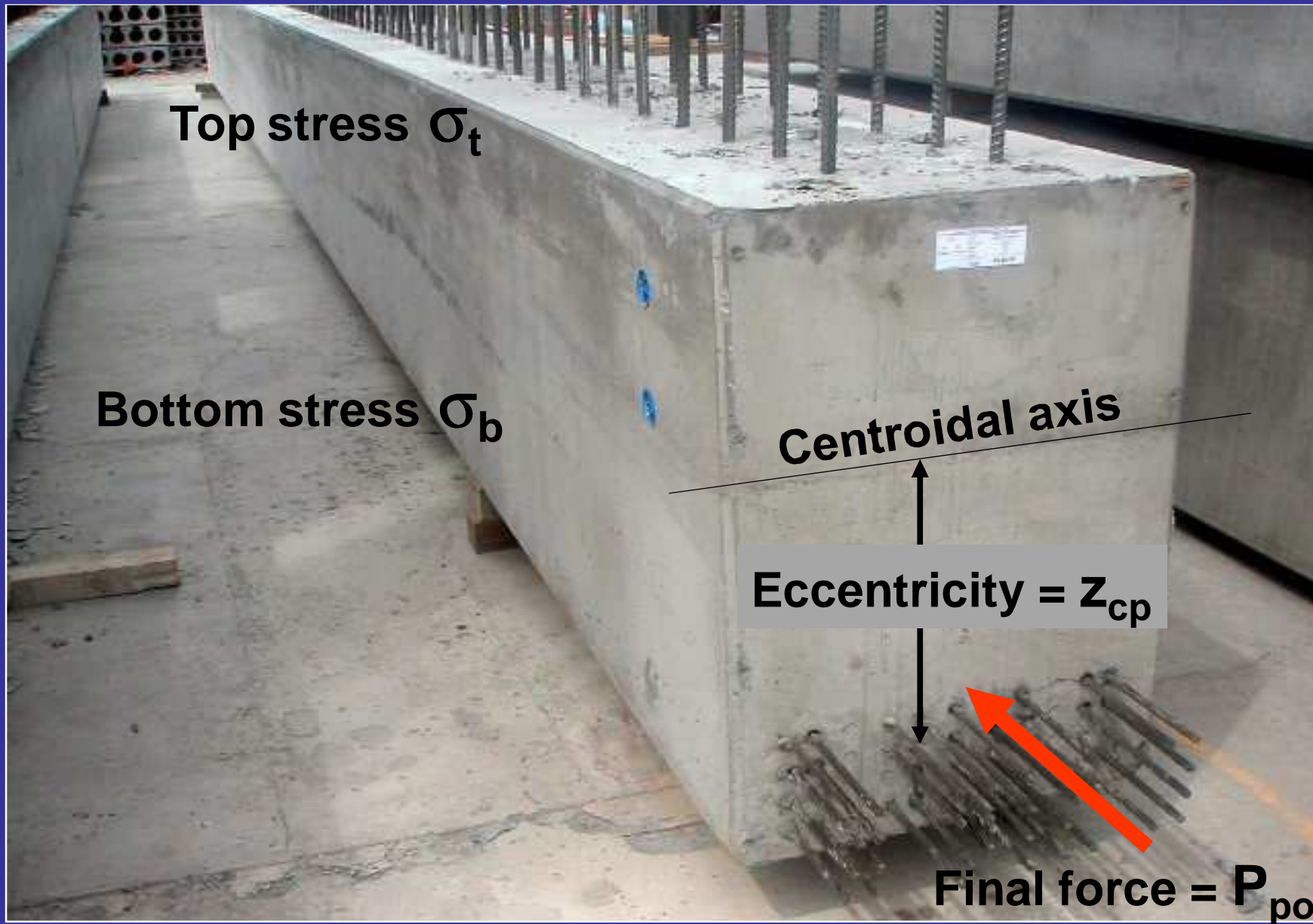
**Prestress.  
Losses.  
Limit  $f_{ctm}$   
Moment of  
resistance.**

**Camber.  
Creep.  
Deflections.  
Limits.**

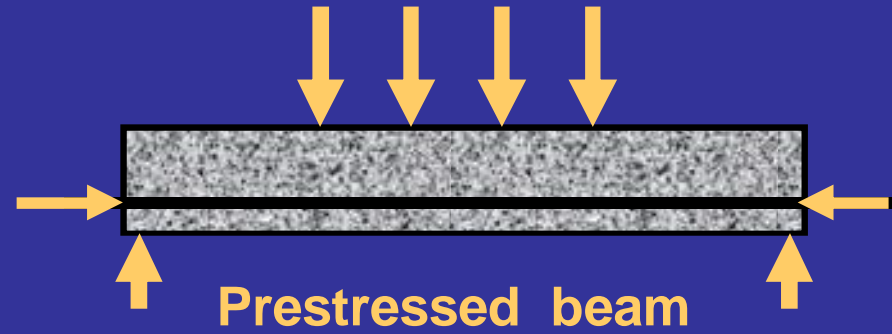
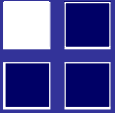
**Ultimate  
strength.  
Equilibrium.  
Compatibility  
 $M_{Rd}$**



## Serviceability stress check

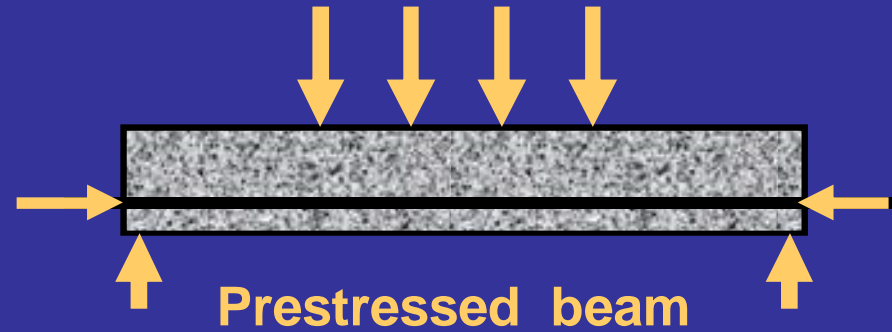
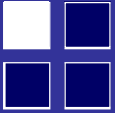






Load or  
moment

Deflection

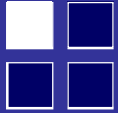


Load or  
moment



-ve camber

Deflection



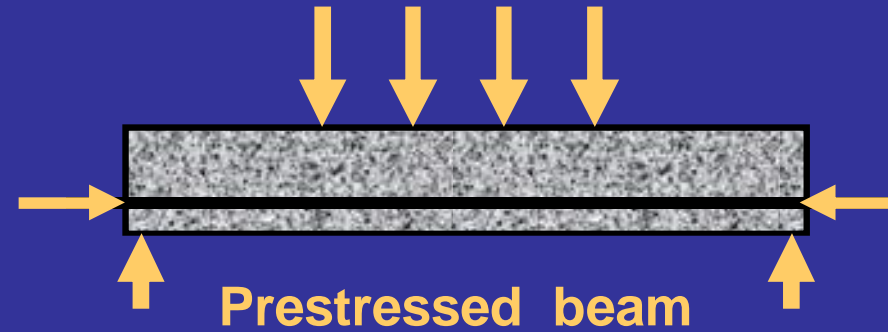
Cracking occurs where  
 $+σ_b - M / Z_B > - f_{ct}$

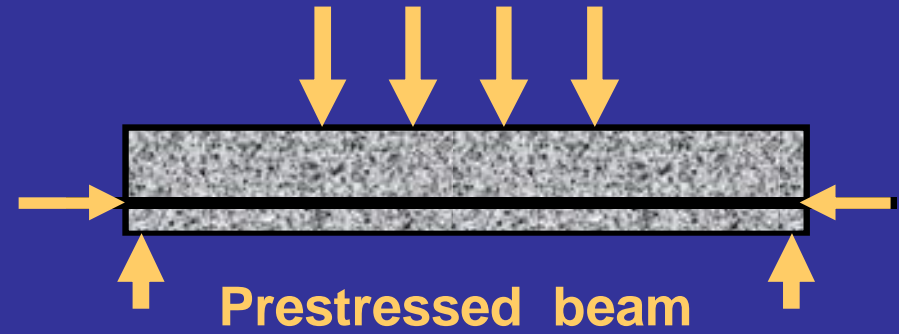
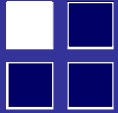
Load or  
moment



-ve camber

Deflection





Load or moment



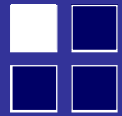
$M_{Rd}$

Service moment of resistance

Ultimate moment of resistance

-ve camber

Deflection

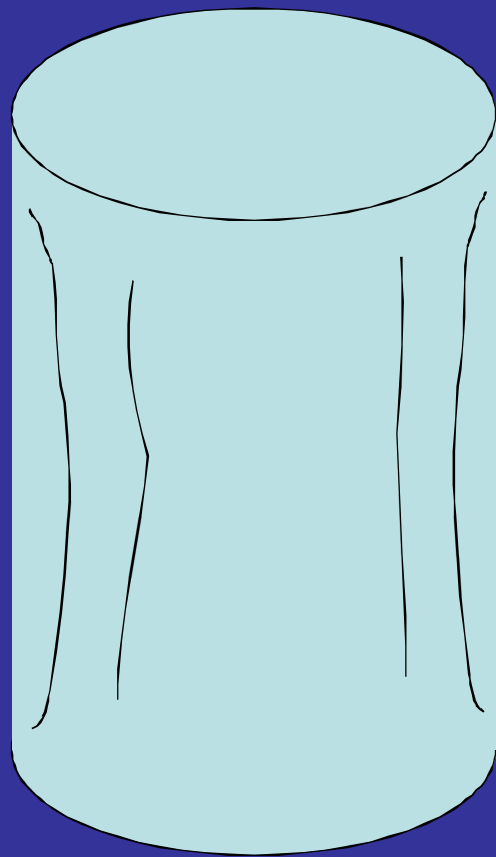


## Compressive cylinder strength $f_{ck}$

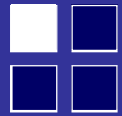
Design compressive strength

$$f_{cd} = 0.85 f_{ck} / \gamma_c$$

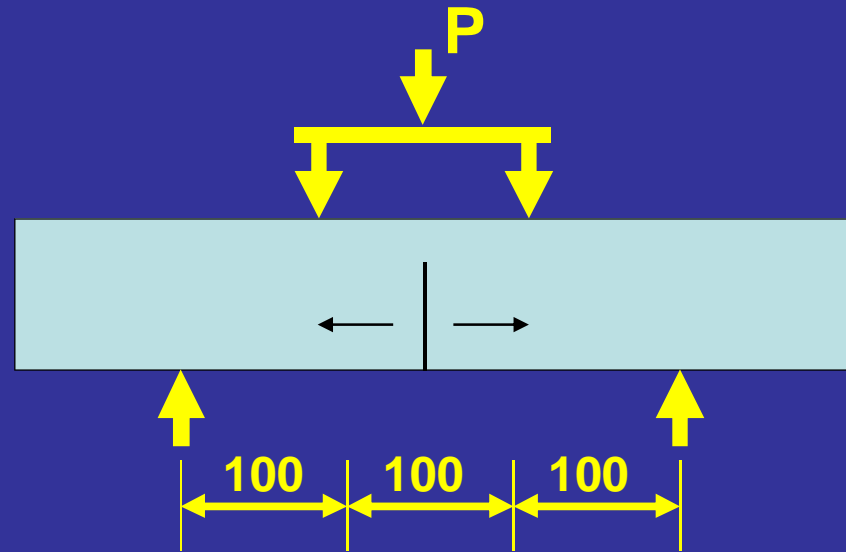
for example  $0.85 \times 45 / 1.5 = 25.5 \text{ N/mm}^2$



Central region  
under uniform  
compressive  
stress



## Tensile strength due to flexure $f_{ctm}$

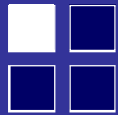


EC2-1-1, Table 3.1

$$f_{ctm} = 0.3 f_{ck}^{2/3}$$

$$\text{e.g.} = 0.3 \times 45^{2/3} = 3.80 \text{ N/mm}^2$$





## EC2-1-1 values listed in Table 3.1

Strength classes for concrete														
$f_{ck}$ (MPa)	12	16	20	25	30	35	40	45	50	55	60	70	80	90
$f_{ck,cube}$ (MPa)	15	20	25	30	37	45	50	55	60	67	75	85	95	105
$f_{cm}$ (MPa)	20	24	28	33	38	43	48	53	58	63	68	78	88	98
$f_{ctm}$ (MPa)	1,6	1,9	2,2	2,6	2,9	3,2	3,5	3,8	4,1	4,2	4,4	4,6	4,8	5,0
$f_{ctk, 0,05}$ (MPa)	1,1	1,3	1,5	1,8	2,0	2,2	2,5	2,7	2,9	3,0	3,1	3,2	3,4	3,5
$f_{clk, 0,95}$ (MPa)	2,0	2,5	2,9	3,3	3,8	4,2	4,6	4,9	5,3	5,5	5,7	6,0	6,3	6,6
$E_{cm}$ (GPa)	27	29	30	31	33	34	35	36	37	38	39	41	42	44

Mean  $f_{cm} = f_{ck} + 8 \text{ N/mm}^2$

## Early tensile stress at transfer

### EC2-1-1, clause 3.1.2.

(9) The development of tensile strength with time is strongly influenced by curing and drying conditions as well as by the dimensions of the structural members. As a first approximation it may be assumed that the tensile strength  $f_{ctm}(t)$  is equal to:

$$f_{ctm}(t) = (\beta_{cc}(t))^{\alpha} \cdot f_{ctm} \quad (3.4)$$

where  $\beta_{cc}(t)$  follows from Expression (3.2) and

$$\alpha = 1 \text{ for } t < 28$$

$$\alpha = 2/3 \text{ for } t \geq 28. \text{ The values for } f_{ctm} \text{ are given in Table 3.1.}$$

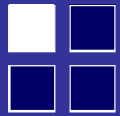
(6) The compressive strength of concrete at an age  $t$  depends on the type of cement, temperature and curing conditions. For a mean temperature of 20°C and curing in accordance with EN 12390 the compressive strength of concrete at various ages  $f_{cm}(t)$  may be estimated from Expressions (3.1) and (3.2).

$$f_{cm}(t) = \beta_{cc}(t) f_{cm} \quad (3.1)$$

$$\therefore f_{ctm}(t) = f_{ctm} \times f_{cm}(t) / f_{cm}$$

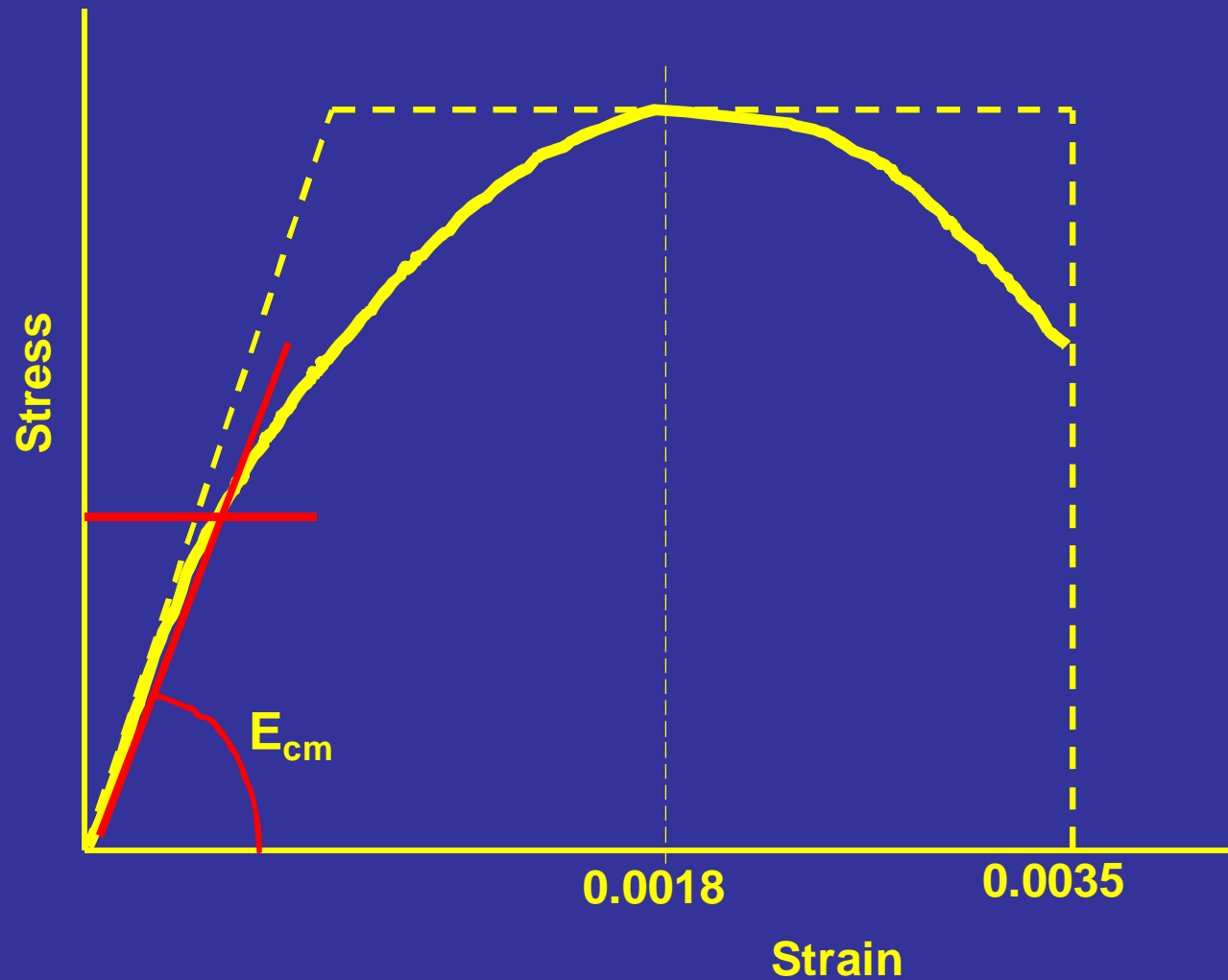
$$\text{e.g. } f_{ctm}(t) = 3.80 \times 38 / 53 = 2.72 \text{ N/mm}^2$$

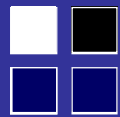




$E_{cm} = 22 (f_{cm} / 10)^{0.3}$  kN/mm<sup>2</sup> for gravel and granite.  
For limestone x 0.9.

e.g. =  $22 \times (53/10)^{0.3} = 36.3$  kN/mm<sup>2</sup>





## Prestressing tendons stress v strain

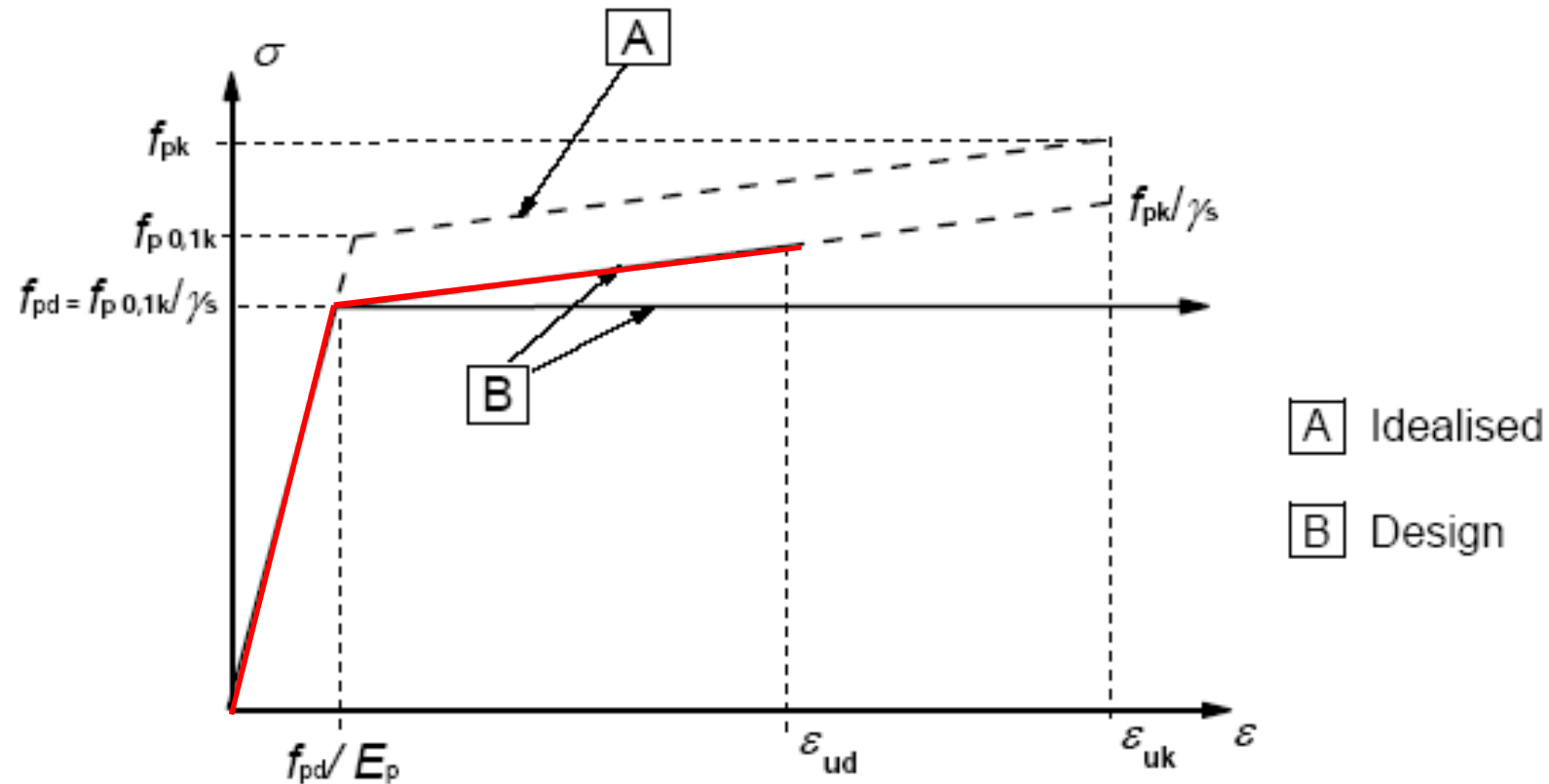
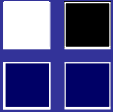


Figure 3.10: Idealised and design stress-strain diagrams for prestressing steel (absolute values are shown for tensile stress and strain)



e.g.  $f_{pk} = 1770 \text{ N/mm}^2$

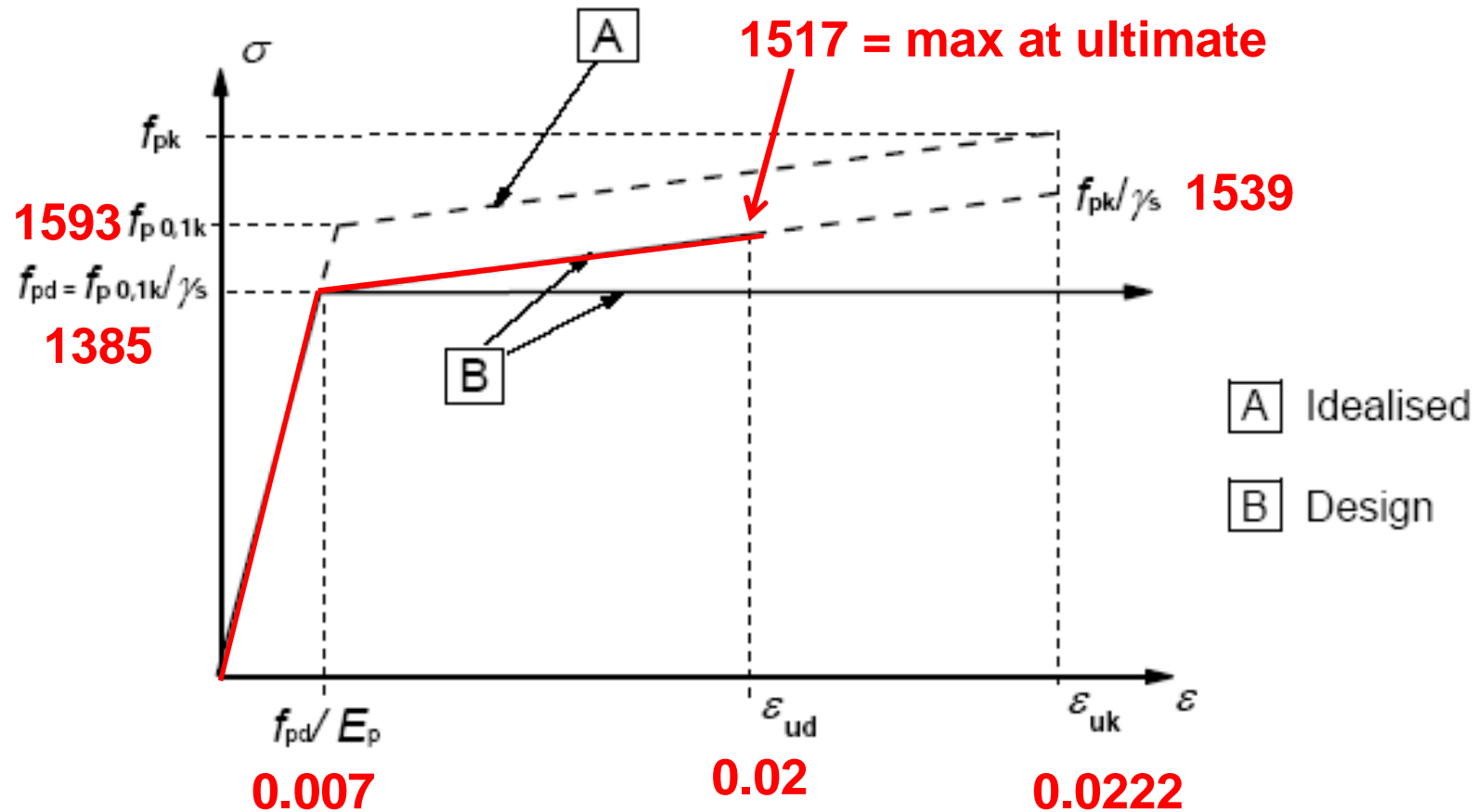


Figure 3.10: Idealised and design stress-strain diagrams for prestressing steel (absolute values are shown for tensile stress and strain)

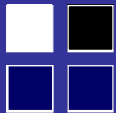
**Durability = nominal cover  $c = c_{\min,dur} + \Delta c_{dev}$**

**For precast slabs > C30/37, use S2**

**Table 4.5N: Values of minimum cover,  $c_{\min,dur}$ , requirements with regard to durability for prestressing steel**

<b>Environmental Requirement for <math>c_{\min,dur}</math> (mm)</b>							
<b>Structural Class</b>	<b>Exposure Class according to Table 4.1</b>						
	X0	XC1	XC2 / XC3	XC4	XD1 / XS1	XD2 / XS2	XD3 / XS3
S1	10	15	20	25	30	35	40
S2	10	15	25	30	35	40	45
S3	10	20	30	35	40	45	50
S4	10	25	35	40	45	50	55
S5	15	30	40	45	50	55	60
S6	20	35	45	50	55	60	65

**Much better and recent information in BS 8500-1: 2015**



## Construction deviation $\Delta c_{dev} = 5 \text{ mm}$ if tendon positions are controlled

### 4.4.1.3 Allowance in design for deviation

(1)P To calculate the nominal cover,  $c_{nom}$ , an addition to the minimum cover shall be made in design to allow for the deviation ( $\Delta c_{dev}$ ). The required minimum cover shall be increased by the absolute value of the accepted negative deviation.

**Note:** The value of  $\Delta c_{dev}$  for use in a Country may be found in its National Annex. The recommended value is 10 mm.

(3) In certain situations, the accepted deviation and hence allowance,  $\Delta c_{dev}$ , may be reduced.

**Note:** The reduction in  $\Delta c_{dev}$  in such circumstances for use in a Country may be found in its National Annex. The recommended values are:

- where fabrication is subjected to a quality assurance system, in which the monitoring includes measurements of the concrete cover, the allowance in design for deviation  $\Delta c_{dev}$  may be reduced:

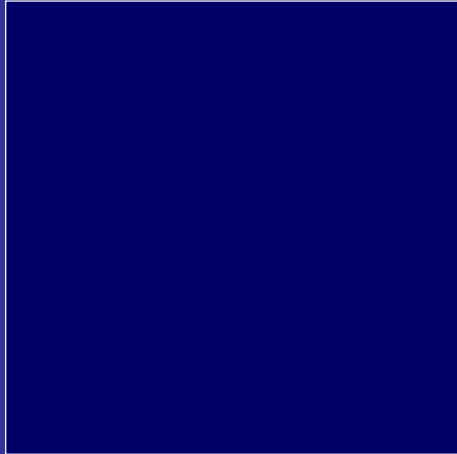
$$10 \text{ mm} \geq \Delta c_{dev} \geq 5 \text{ mm}$$

(4.3N)

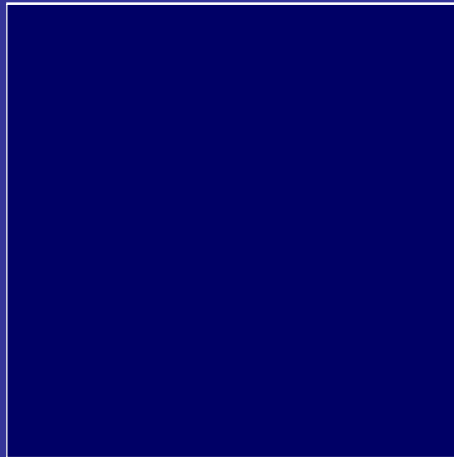
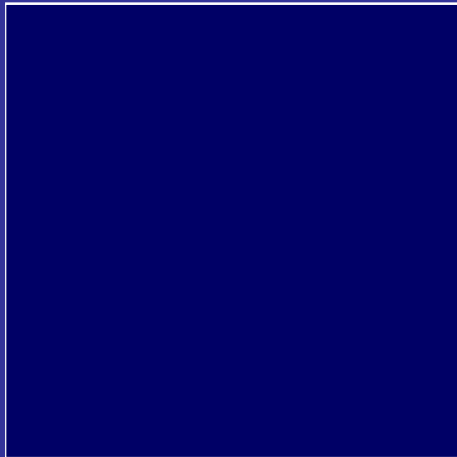


# Syllabus

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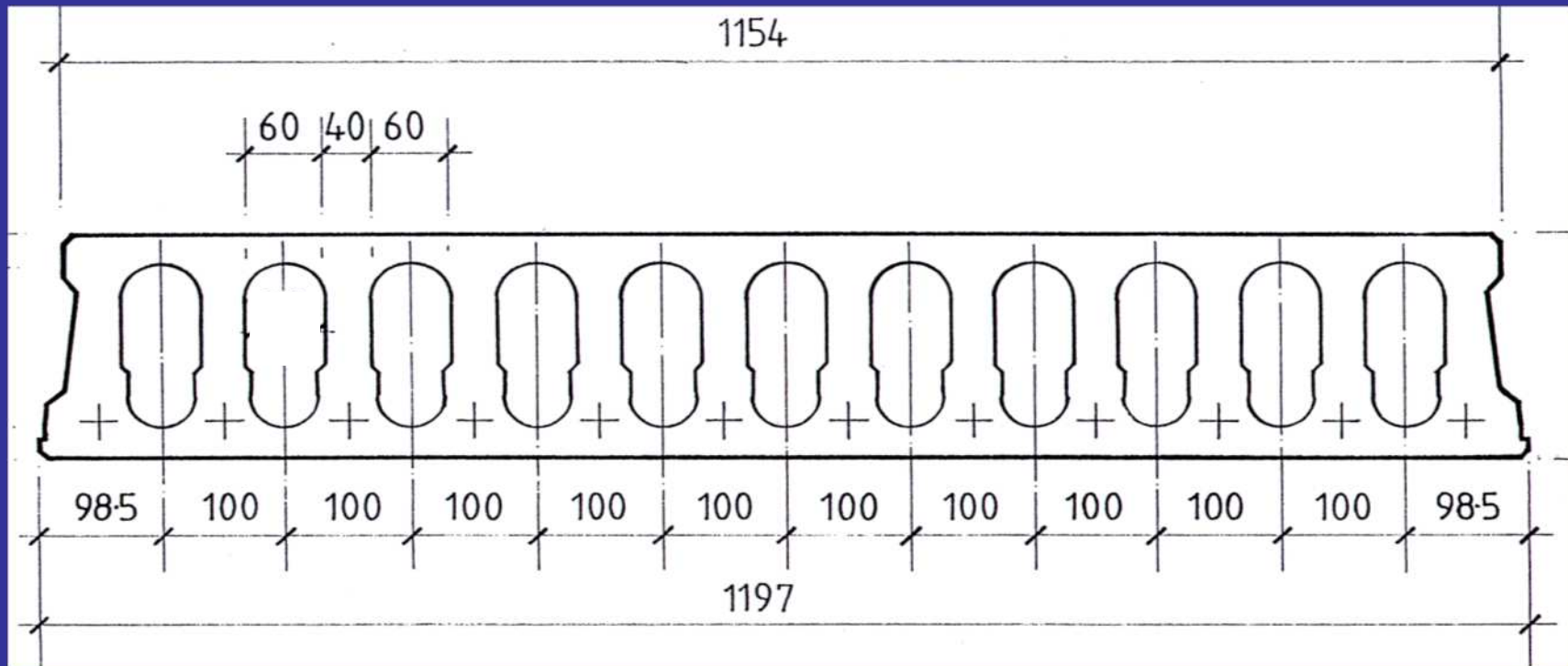
**Prestress.  
Losses.  
Limit  $f_{ctm}$   
Moment of  
resistance.**



## Prestressed concrete hollow core floor units







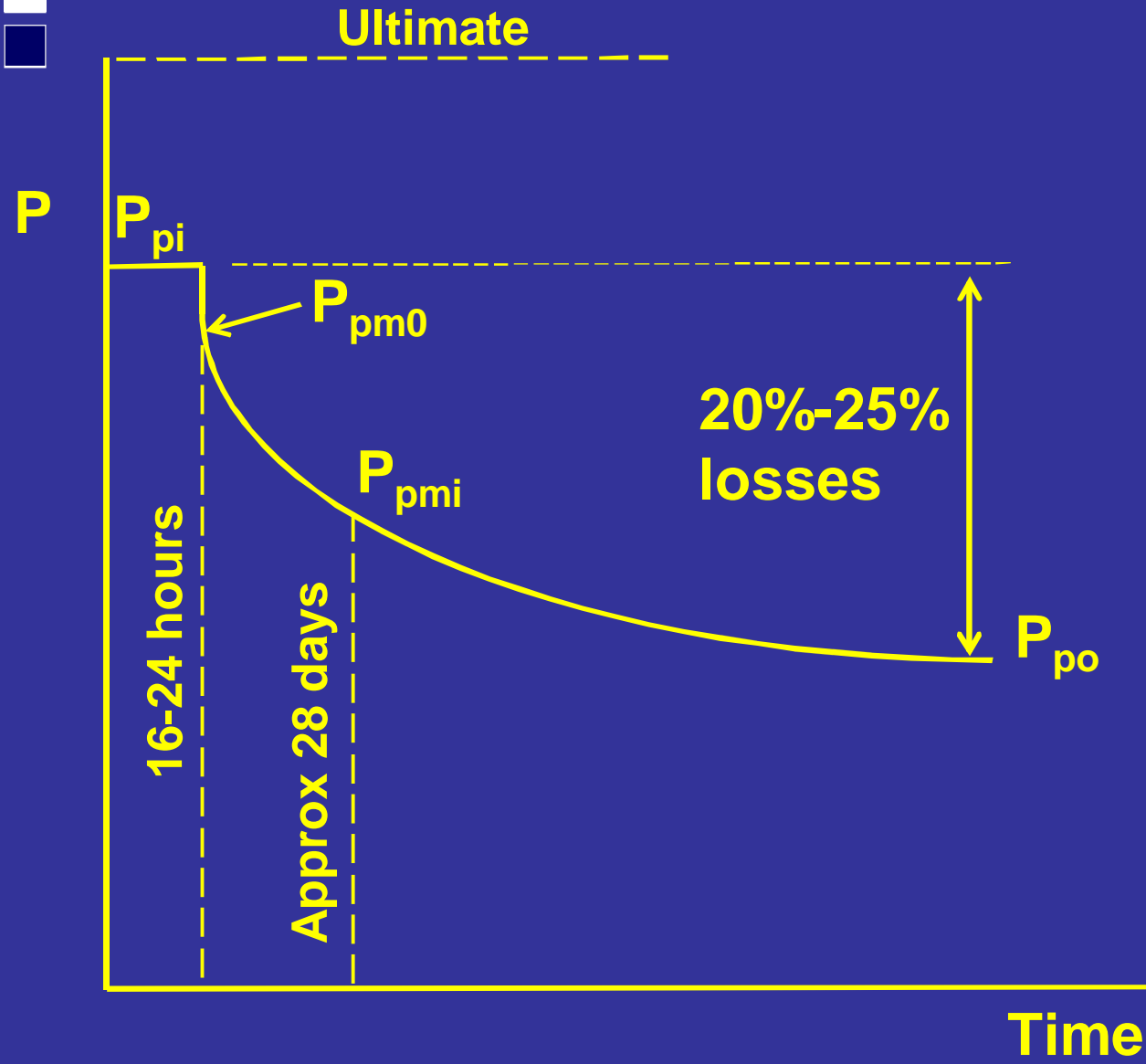
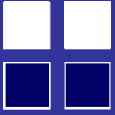
**Calculate: Area  $A_c$**

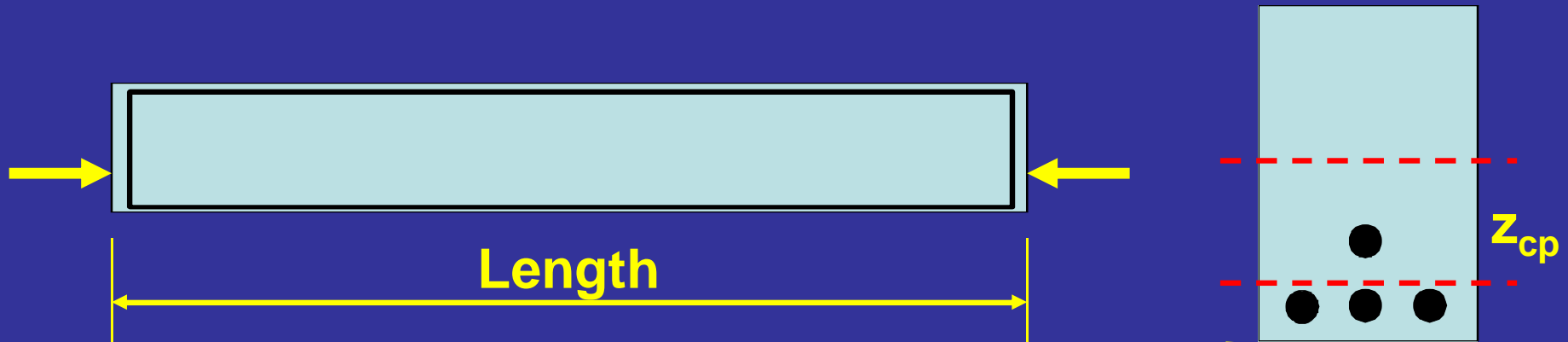
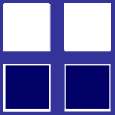
**Centroid height  $y_b$  (and  $y_t = h - y_b$ )**

**2<sup>nd</sup> moment of area  $I_{x-x}$**

**1<sup>st</sup> moment of area  $S_{x-x}$  (for shear only)**

**Section modulus  $Z_b = I_{x-x} / y_b$  and  $Z_T = I_{x-x} / y_t$**



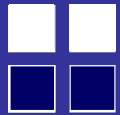


Initial prestressing force  $P_{pi} = \eta f_{pk} A_{ps}$

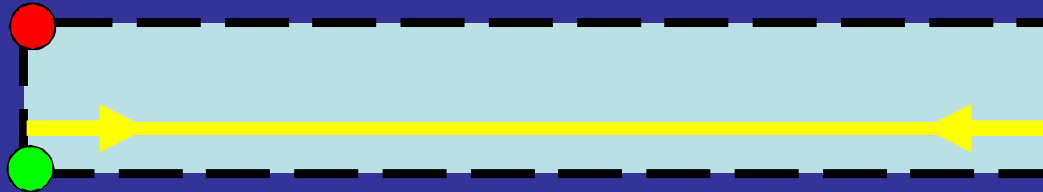
Eccentricity  $z_{cp}$

Typically  $0.7 \times 1770$

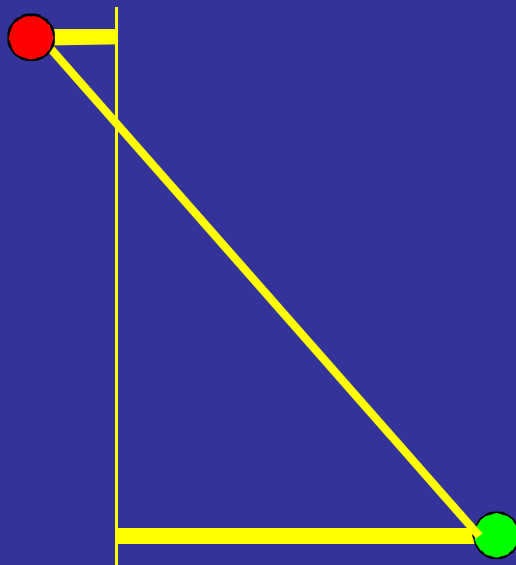
Immediate strand relaxation and elastic shortening at transfer ...



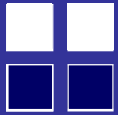
Transfer stress limits, e.g. based on  $f_{ck}(t) = 30 \text{ N/mm}^2$



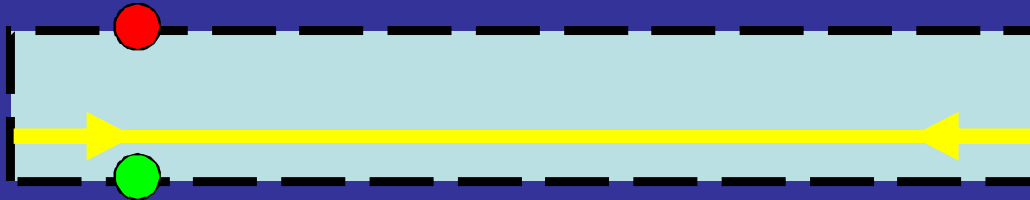
$$\sigma_{t(t)} \geq -f_{ctm}(t) = -2.72 \text{ N/mm}^2$$



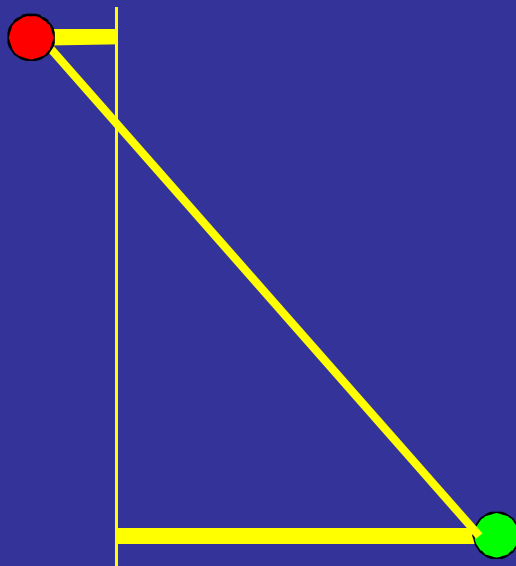
$$\sigma_{b(t)} \leq 0.6 f_{ck}(t) = +18.00 \text{ N/mm}^2$$



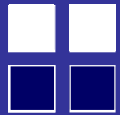
Can be checked at the end of the transmission length  $l_{pt}$



$$\sigma_{t(t)} + M_{\text{self}} / Z_t \geq - f_{ctm}(t) = - 2.72 \text{ N/mm}^2$$

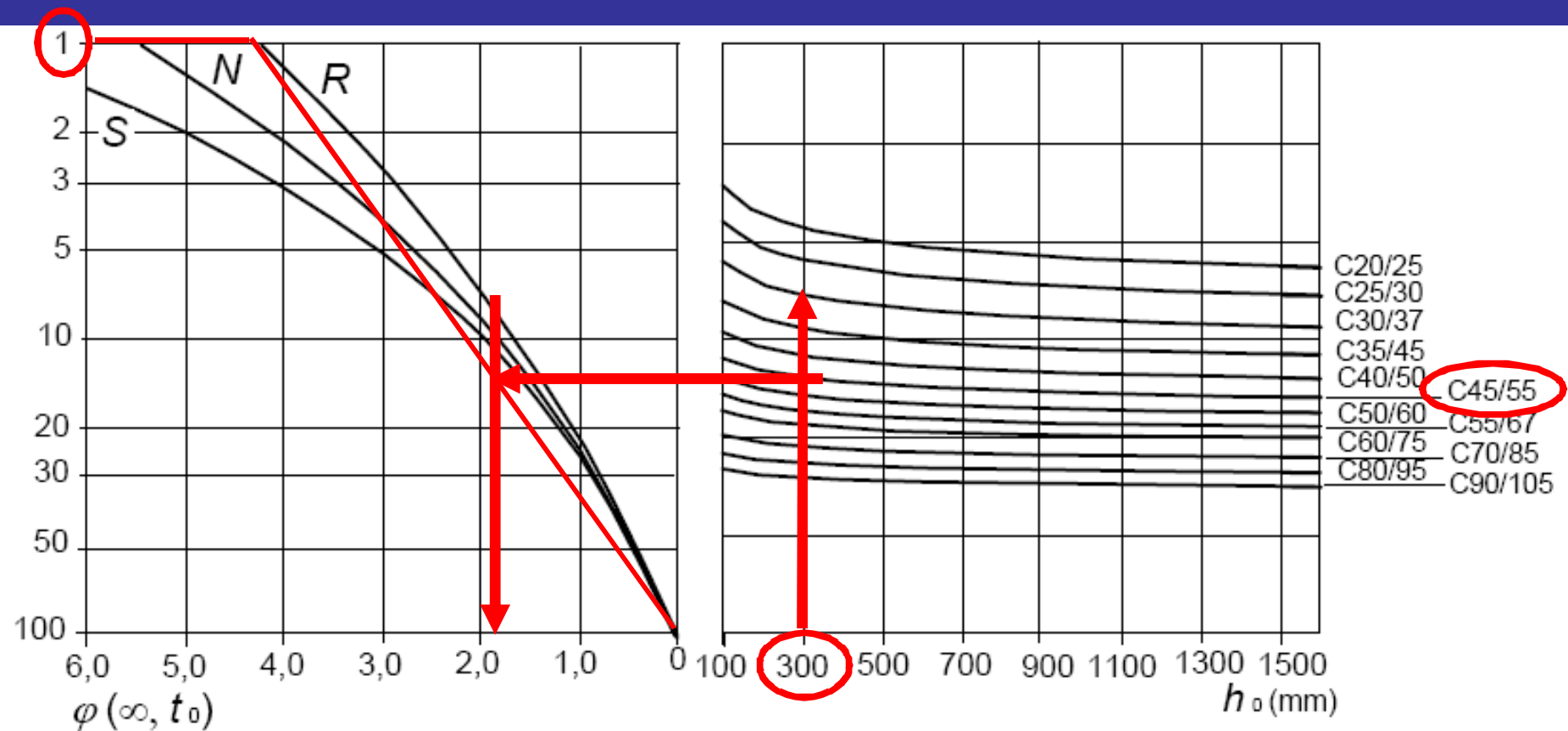


$$\sigma_{b(t)} - M_{\text{self}} / Z_b \leq 0.6 f_{ck}(t) = +18.00 \text{ N/mm}^2$$

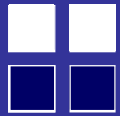


Followed by long term losses due to creep, shrinkage and further strand relaxation

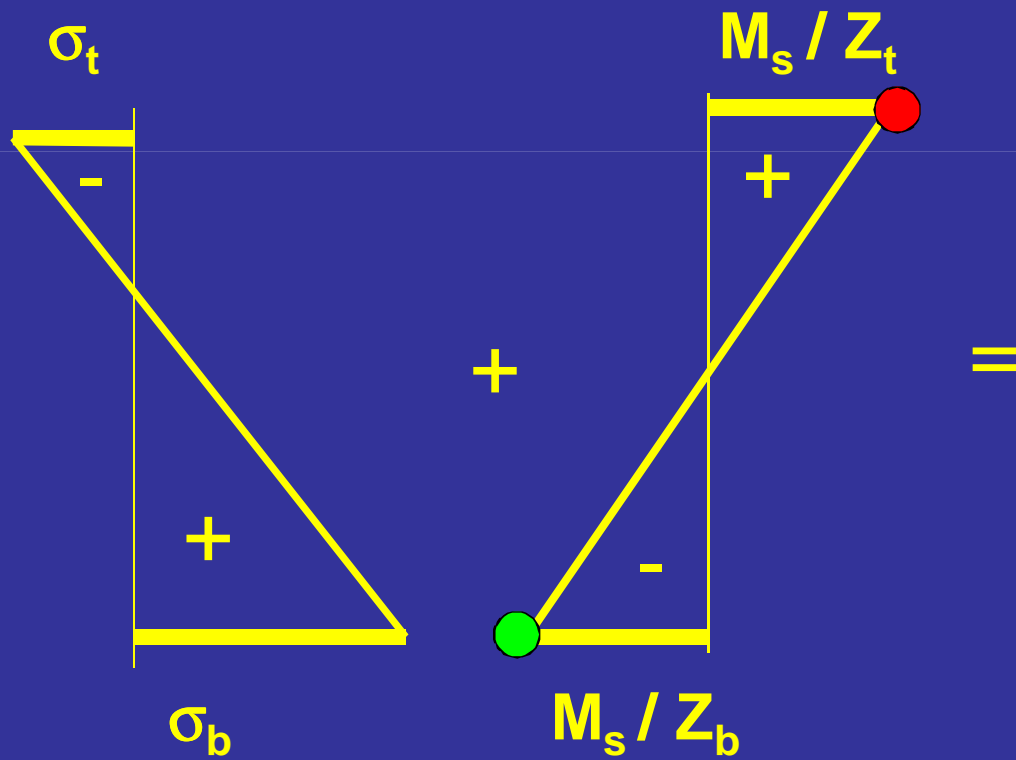
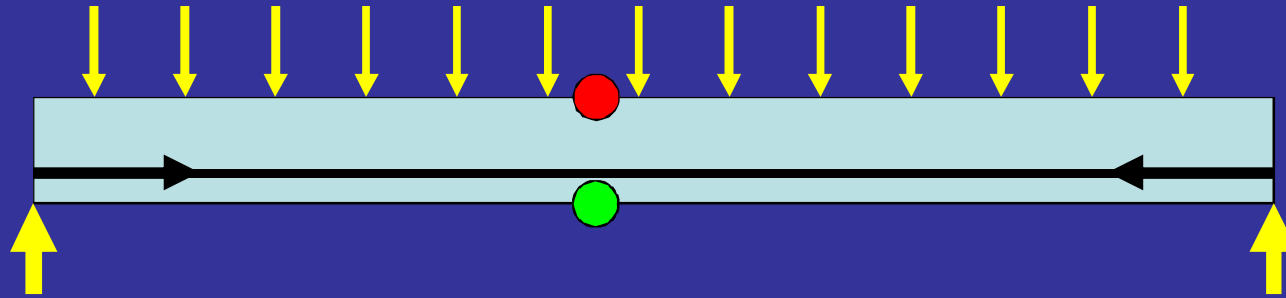
Results in a lower prestress  $\sigma_b$  and  $\sigma_t$

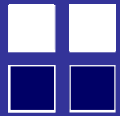


b) outside conditions - RH = 80%

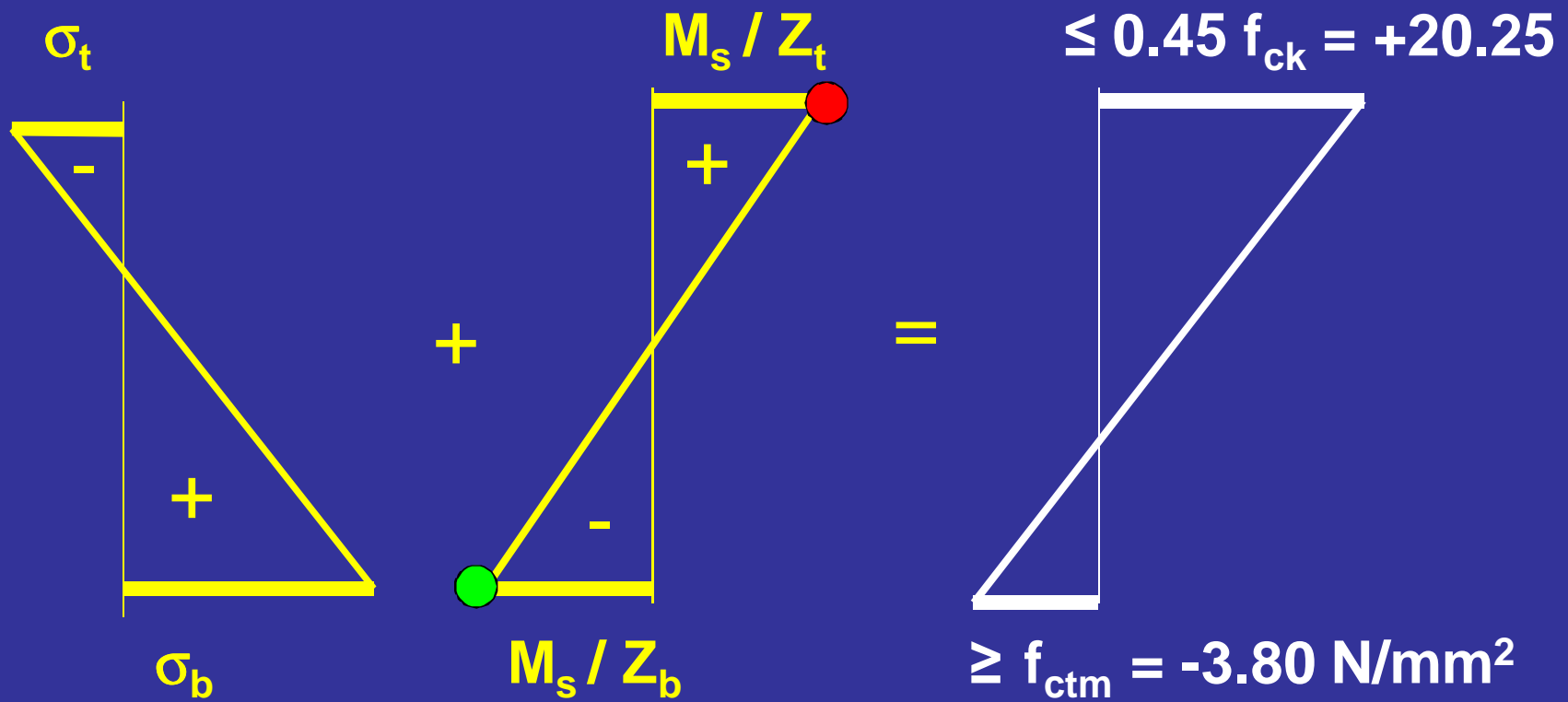
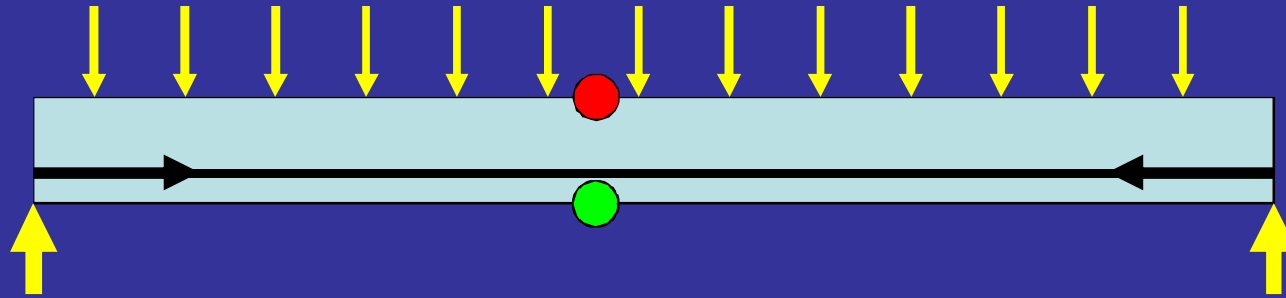


# Adding prestress to imposed service stress =

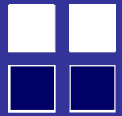




# Adding prestress to imposed service stress =

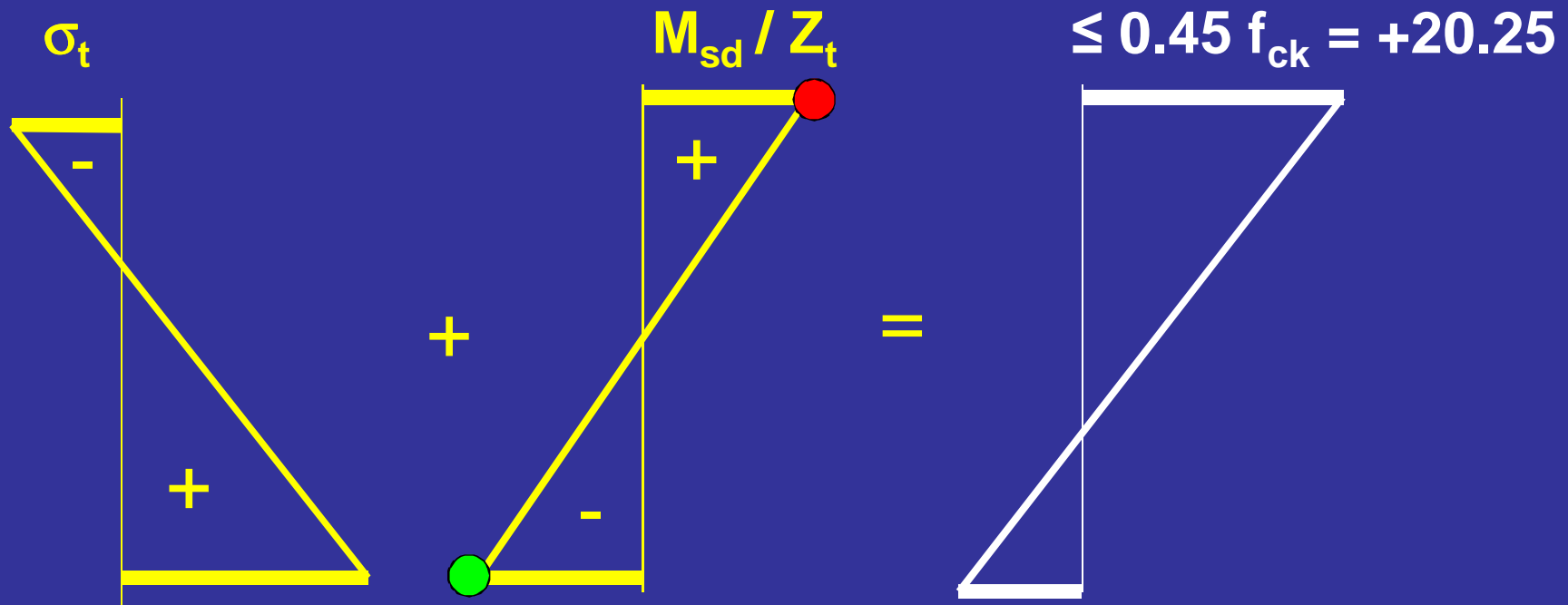


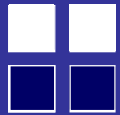




Service moment of resistance is lesser of

$$M_{Sd} = (\sigma_t + 0.45f_{ck}) Z_t$$

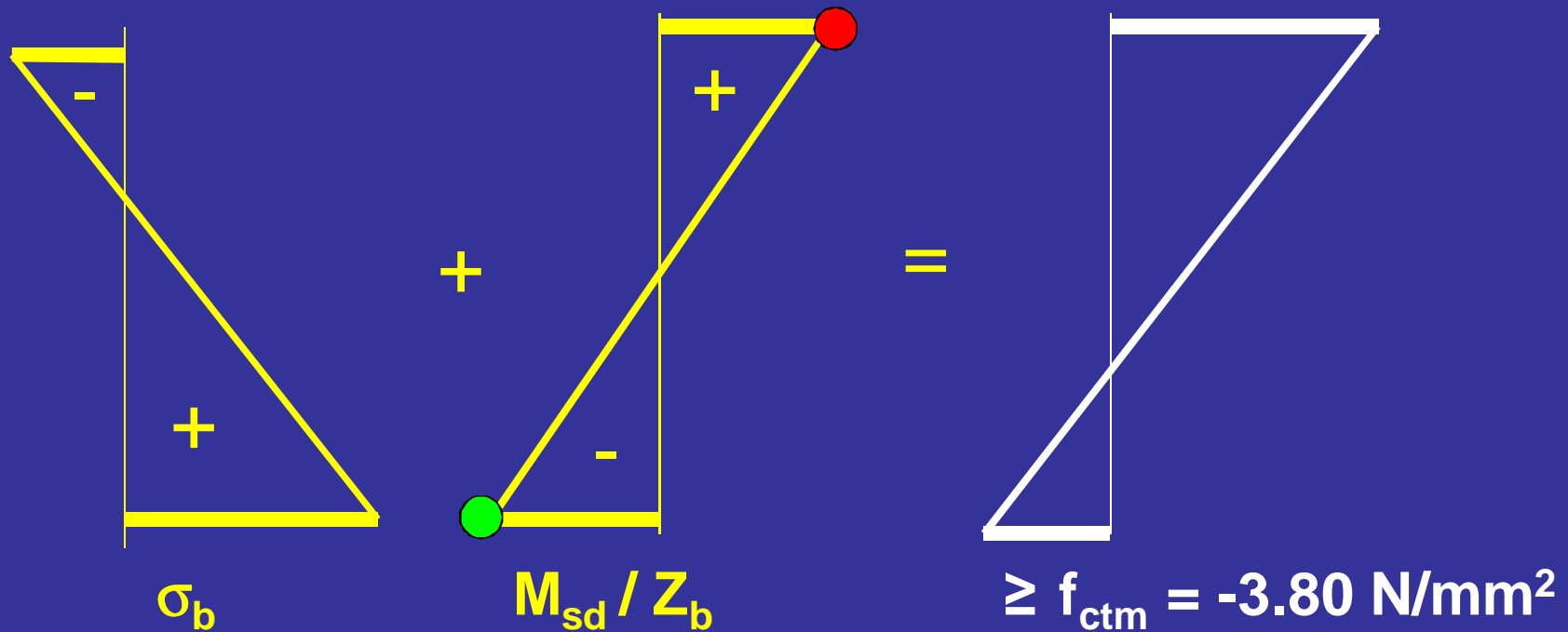


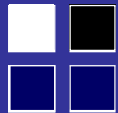


## Service moment of resistance is lesser of

$$M_{Sd} = (\sigma_t + 0.45f_{ck}) Z_t$$

$$M_{Sd} = (\sigma_b + f_{ctm}) Z_b \quad \text{Mostly critical}$$





**But for exposure > XC1, permissible tension  $f_{ctm}$  is reduced (depending on each country)**

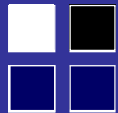
(5) A limiting calculated crack width,  $w_{max}$ , taking into account the proposed function and nature of the structure and the costs of limiting cracking, should be established.

**Note:** The value of  $w_{max}$  for use in a Country may be found in its National Annex. The recommended values for relevant exposure classes are given in Table 7.1N.

Table 7.1N Recommended values of  $w_{max}$  (mm)

**UK National Annex**

Exposure Class	Reinforced members and prestressed members with unbonded tendons	Prestressed members with bonded tendons
	Quasi-permanent load combination	Frequent load combination
X0, XC1	0,4 <sup>1</sup>	0,2
XC2, XC3, XC4	0,3	0,2 <sup>2</sup>
XD1, XD2, XS1, XS2, XS3		Decompression
<b>Note 1:</b> For X0, XC1 exposure classes, crack width has no influence on durability and this limit is set to guarantee acceptable appearance. In the absence of appearance conditions this limit may be relaxed.		
<b>Note 2:</b> For these exposure classes, in addition, decompression should be checked under the quasi-permanent combination of loads.		



## For Class XC2-XC4 (external) no tension allowed. But can use quasi-permanent live load $\times \psi_2$

(5) A limiting calculated crack width,  $w_{\max}$ , taking into account the proposed function and nature of the structure and the costs of limiting cracking, should be established.

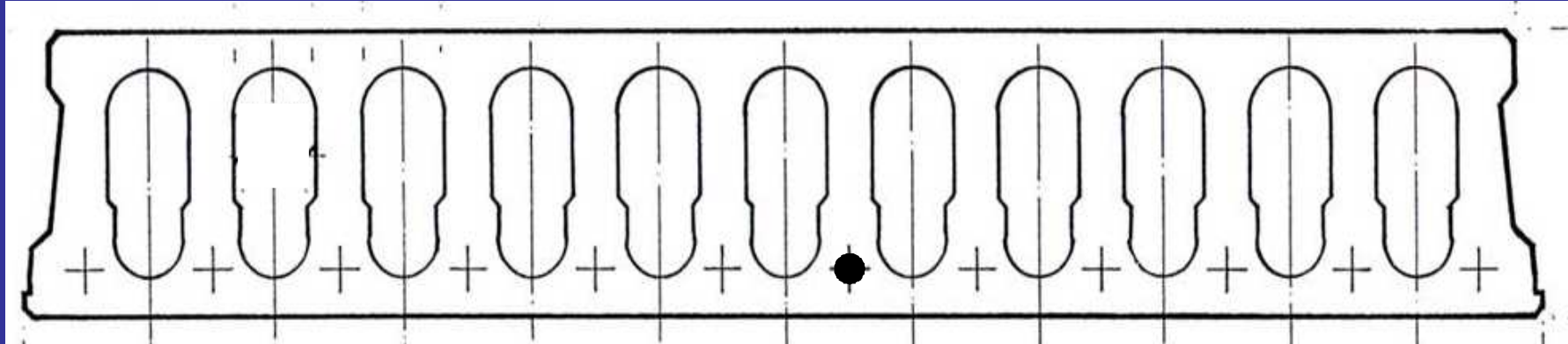
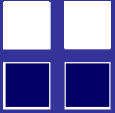
**Note:** The value of  $w_{\max}$  for use in a Country may be found in its National Annex. The recommended values for relevant exposure classes are given in Table 7.1N.

Table 7.1N Recommended values of  $w_{\max}$  (mm)

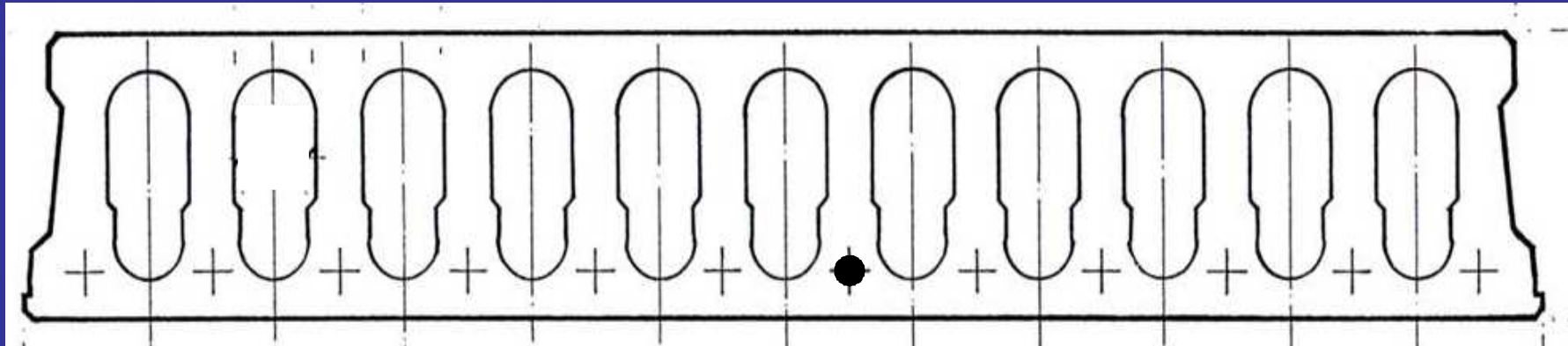
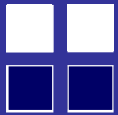
Exposure Class	Reinforced members and prestressed members with unbonded tendons	Prestressed members with bonded tendons
	Quasi-permanent load combination	Frequent load combination
X0, XC1	0,4 <sup>1</sup>	0,2
XC2, XC3, XC4	0,3	0,2 <sup>2</sup>
XD1, XD2, XS1, XS2, XS3		Decompression

**Note 1:** For X0, XC1 exposure classes, crack width has no influence on durability and this limit is set to guarantee acceptable appearance. In the absence of appearance conditions this limit may be relaxed.

**Note 2:** For these exposure classes, in addition, decompression should be checked under the quasi-permanent combination of loads.



**Worked example:**



**Worked example: XC1 exposure**

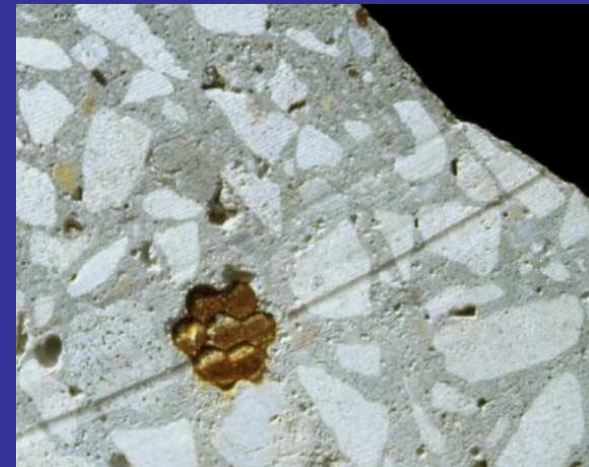
**1200 wide x 250 depth**

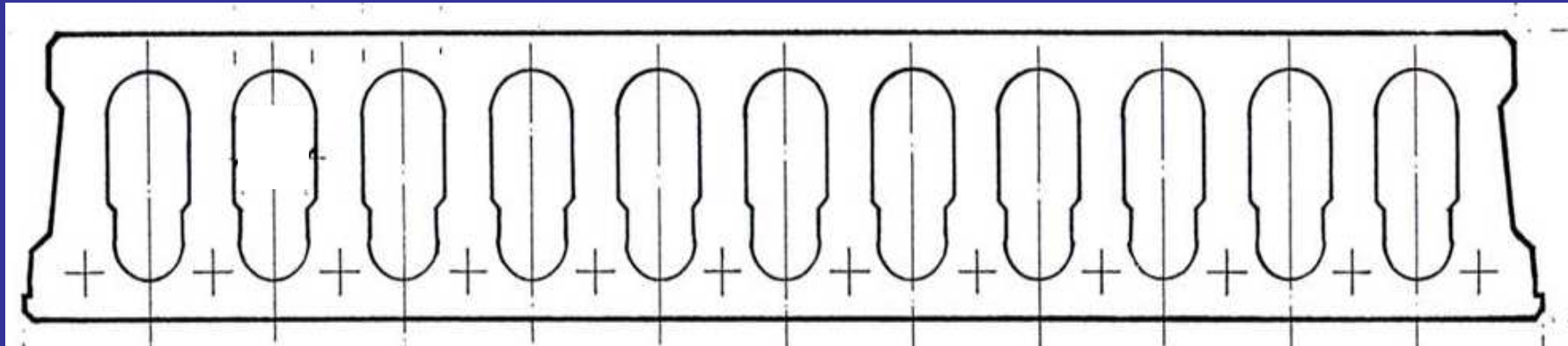
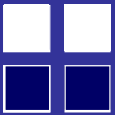
**12 no. 9.3 mm strands at 35 mm cover**

**Axis a = 39.6 mm**

**$f_{pk} = 1770 \text{ N/mm}^2$**

**Initial stressing to 70% =  $1239 \text{ N/mm}^2$**





$$A_c = 182791 \text{ mm}^2$$

$$y_b = 122.4 \text{ mm}$$

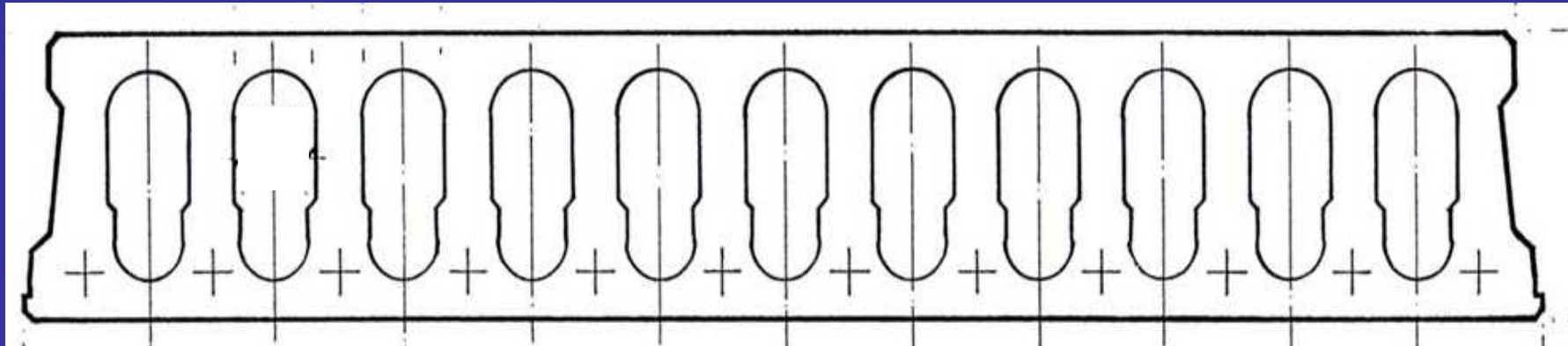
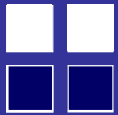
$$I_{x-x} = 1270.3 \times 10^6 \text{ mm}^4$$

$$Z_b = 1270.3 / 122.4 = 10.378 \times 10^6 \text{ mm}^3$$

$$Z_T = 9.955 \times 10^6 \text{ mm}^3$$

$$A_{ps} = 12 \times 52 = 624 \text{ mm}^2$$

$$z_{cp} = 122.4 - 39.6 = 82.8 \text{ mm}$$



**Initial prestress  $P_{pi} = 624 \times 1239 \times 10^{-3} = 773.1 \text{ kN}$**

**Relaxation Class 2. 2.5% at 1000 hours**

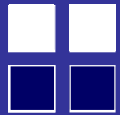
**Immediate relaxation loss =  $4.95 \text{ N/mm}^2$  (0.40%)**

**$E_p$  (strand)  $195 \text{ kN/mm}^2$**

**$E_{cm}(t)$  (gravel aggregate) =  $32.8 \text{ kN/mm}^2$**

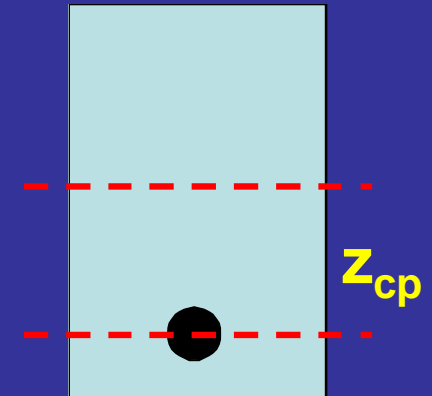
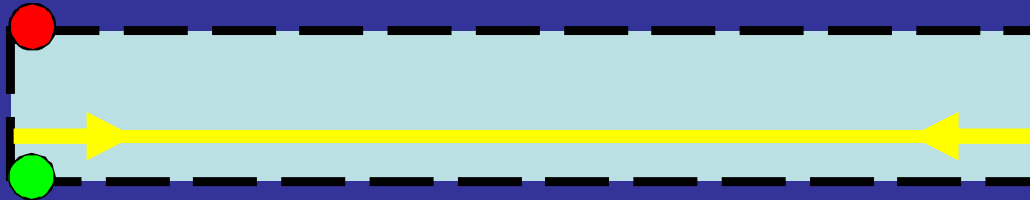
**Elastic shortening loss =  $49.67 \text{ N/mm}^2$  (4.01%)**



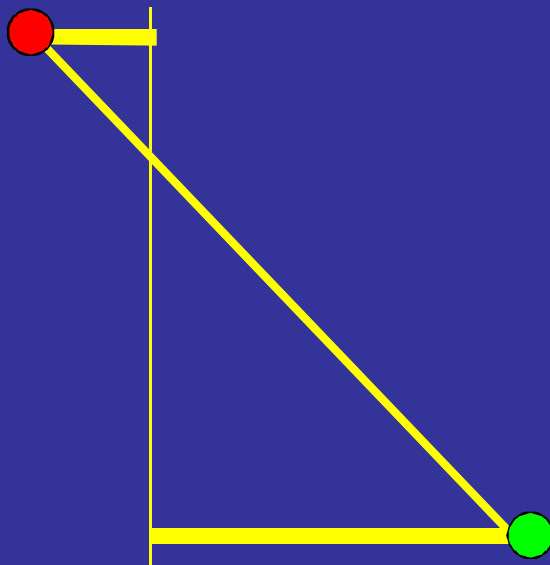


## Remaining force at transfer

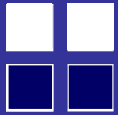
$$P_{pm0} = 739.0 \text{ kN (4.41\% loss)}$$



$$\sigma_{t(t)} = -2.10 \geq -2.72 \text{ N/mm}^2$$



$$\sigma_{b(t)} = 9.94 \leq 18.0 \text{ N/mm}^2 \quad \text{OK}$$



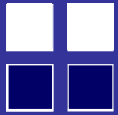
**Further loss of prestress at installation at (say) 28 days:**

**RH at transfer = 70%**

**Creep coefficient at 28 days = 0.84**

**Creep loss of stress at 28 days = 34.0 N/mm<sup>2</sup> (2.74%)**

**$P_{pmi} = 717.8 \text{ kN}$  (this is used later to determine  
camber after installation)**



**Final loss of prestress at 500,000 hours (57 years):**

**RH at service (indoor exposure) = 50%**

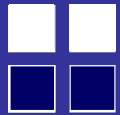
**Creep coefficient from installation to life = 1.60**

**Creep loss of stress = 61.8 N/mm<sup>2</sup> (4.99%)**

**Shrinkage  $\epsilon_{sh} = 420 \times 10^{-6}$**

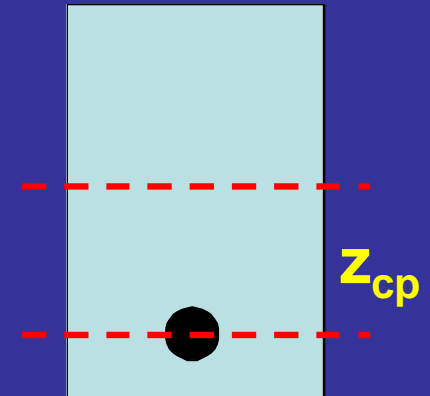
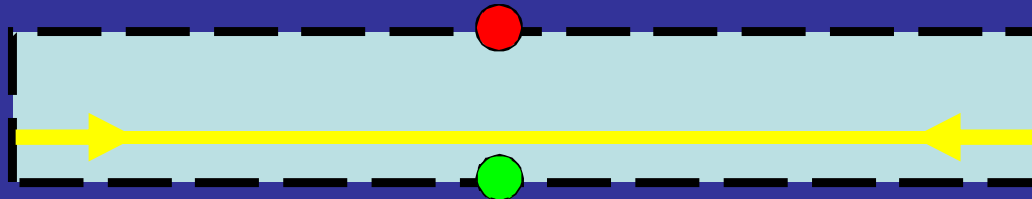
**Shrinkage loss = 74.7 N/mm<sup>2</sup> (6.03%)**

**Long term relaxation loss = 40.3 N/mm<sup>2</sup> (3.25%)**

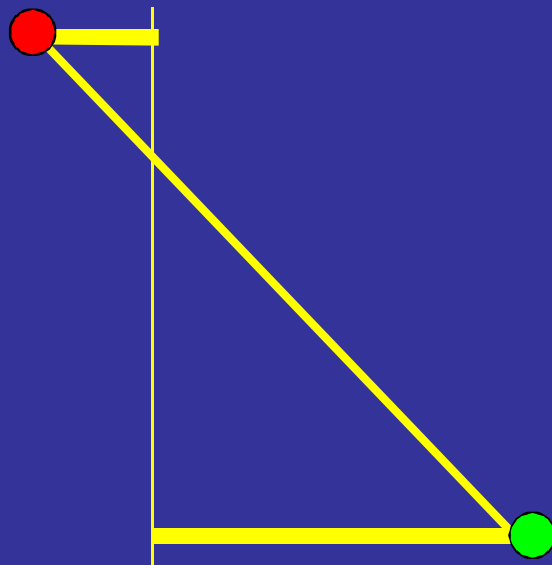


After all losses

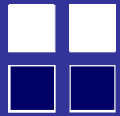
$P_{p0} = 614.3 \text{ kN}$  (20.5% loss)



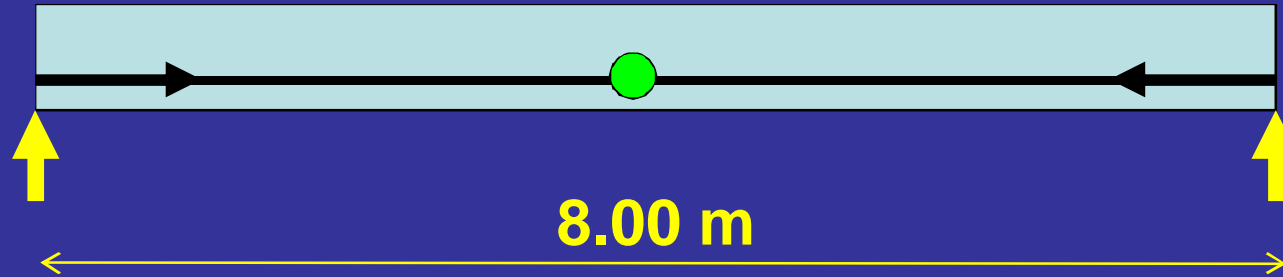
$\sigma_t = -1.746 \text{ N/mm}^2$



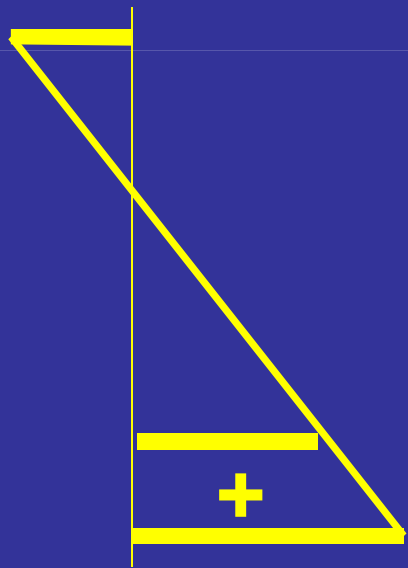
$\sigma_b = 8.26 \text{ N/mm}^2$



**But ! creep losses may be reduced by reversal of prestress due to self weight and dead loads**

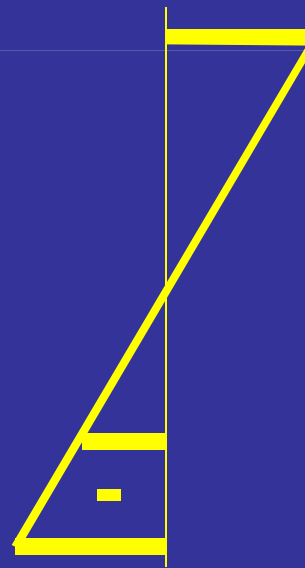


**Self weight  $M = 26.87$  kNm**

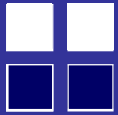


**Creep loss at support  
=  $95.8$  N/mm<sup>2</sup>**

**+**



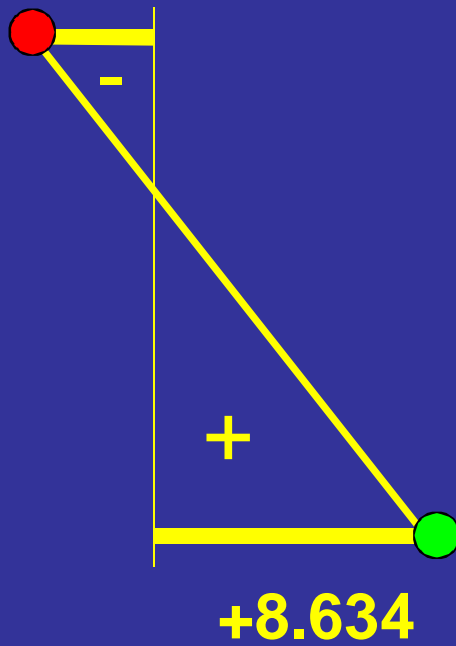
**Creep loss at mid-span  
=  $63.4$  N/mm<sup>2</sup>**

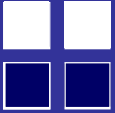


Final  $\sigma_b = 8.634 \text{ N/mm}^2$  (increase of 4.5%)

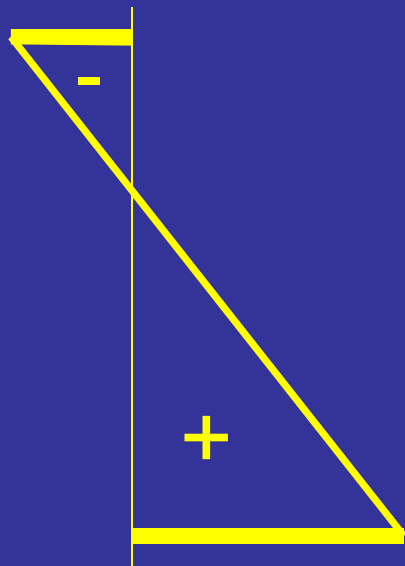
$\sigma_t = -1.825 \text{ N/mm}^2$

= -1.825





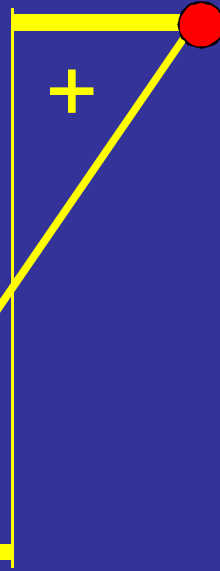
$$= -1.825$$



$$+8.634$$

+

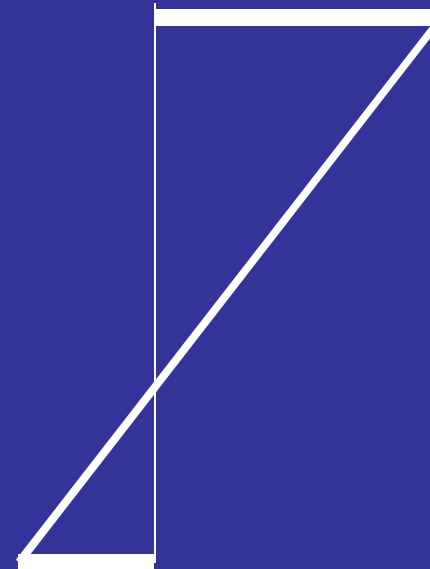
$$M_s / Z_t$$



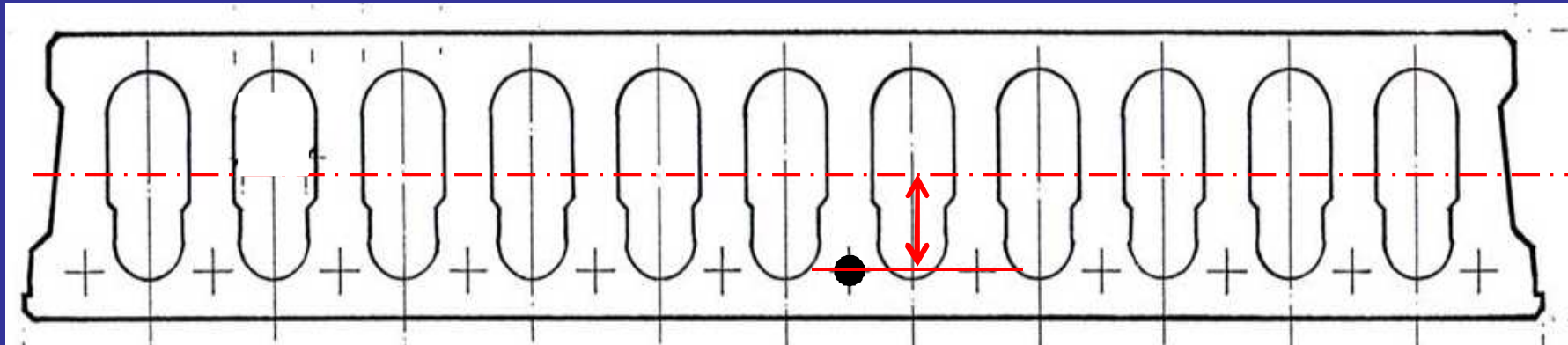
$$M_s / Z_b$$

=

$$\leq 0.45 f_{ck} = +20.25$$



$$\geq f_{ctm} = -3.80 \text{ N/mm}^2$$



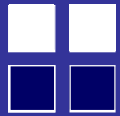
**But ! compound values may be used for  $I_{xx}$ ,  $Z_b$  and  $Z_t$  based on the *transformed area* of the strands:**

$$m = E_p / E_{cm} = (195 / 36.3) - 1 = 4.37$$

$$\text{Then } I_{xx} = 1289 \times 10^6 \text{ mm}^4$$

$$Z_{b,co} = 10.634 \times 10^6 \text{ mm}^3 \text{ (increase of 2.5\%)}$$



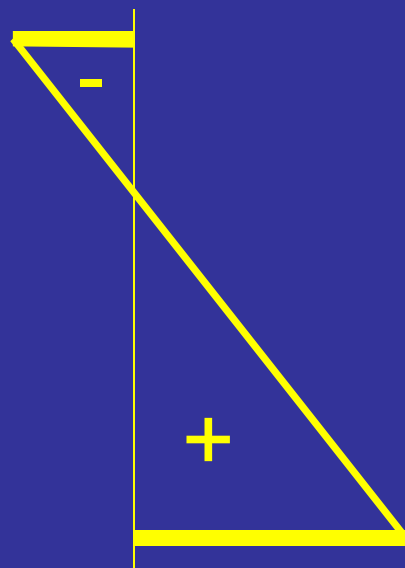


Service moment of resistance is lesser of

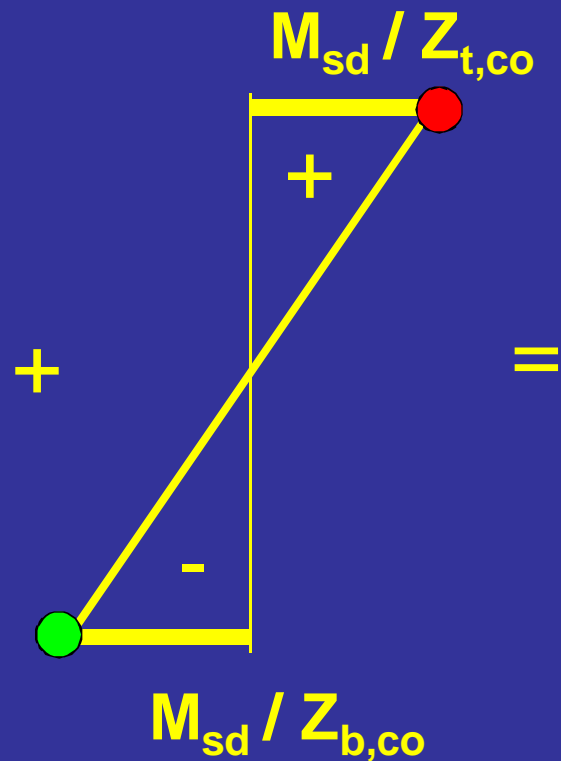
$$M_{Sd} = (20.25 + 1.825) \times 10.004 = 220.8 \text{ kNm}$$

$$M_{Sd} = (8.634 + 3.80) \times 10.634 = \underline{132.2 \text{ kNm}}$$

$$= -1.825$$

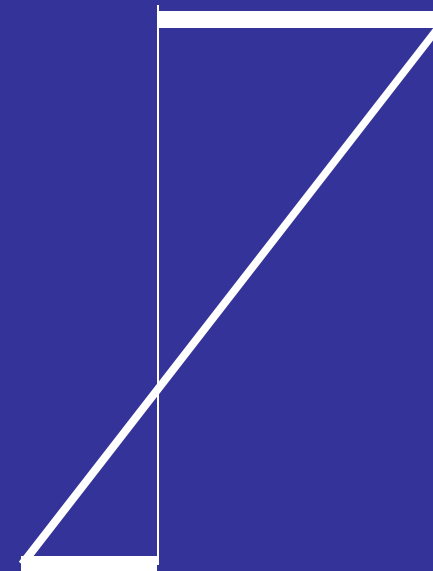


$$+8.634$$

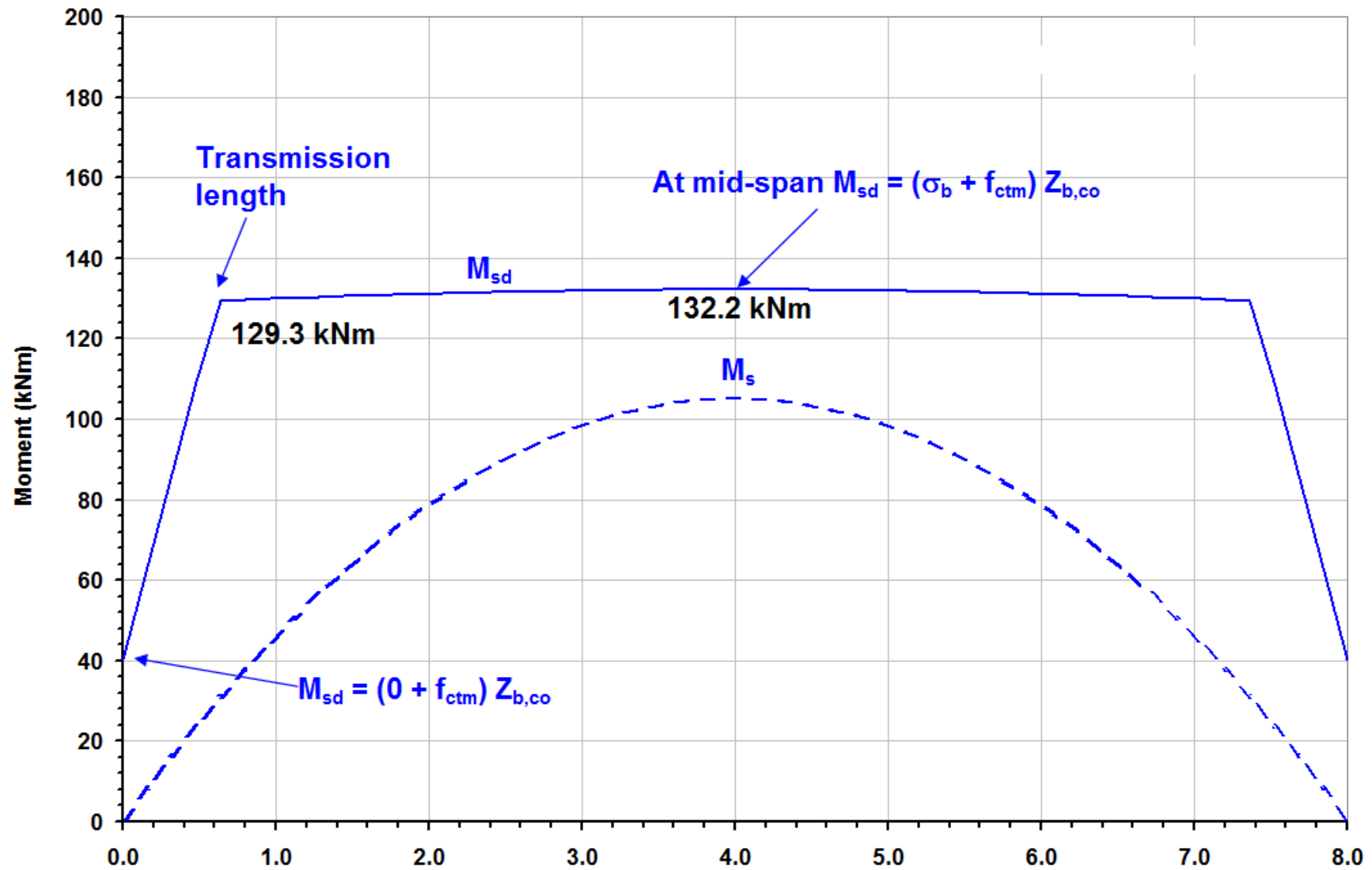


=

$$\leq 0.45 f_{ck} = +20.25$$

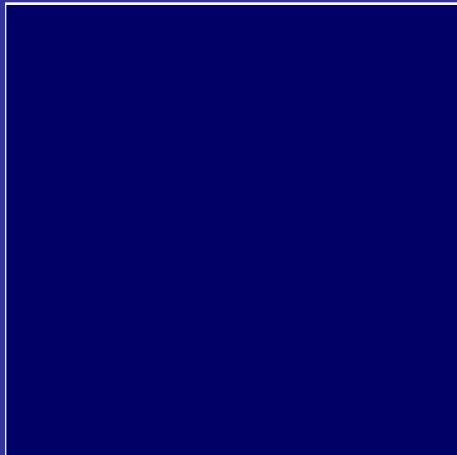
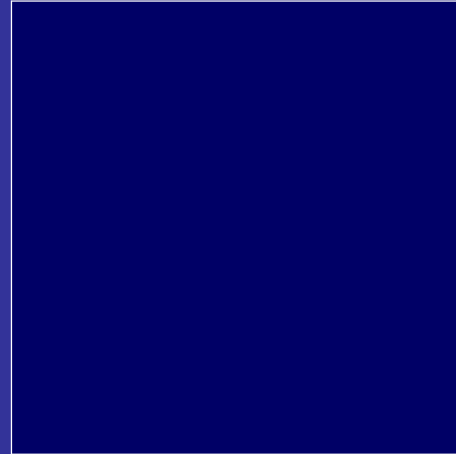
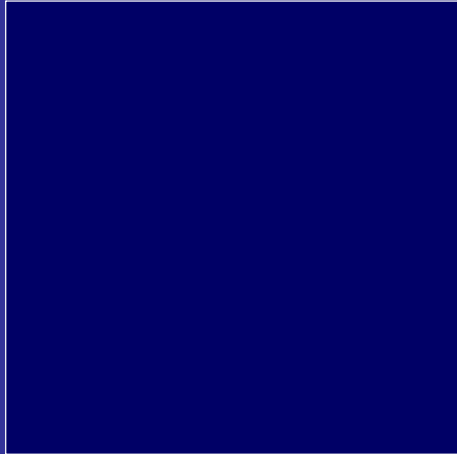


$$\geq f_{ctm} = -3.80 \text{ N/mm}^2$$

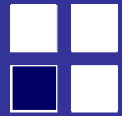


# Syllabus

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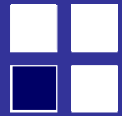


**Ultimate  
strength.  
Equilibrium.  
Compatibility.  
 $M_{Rd}$**

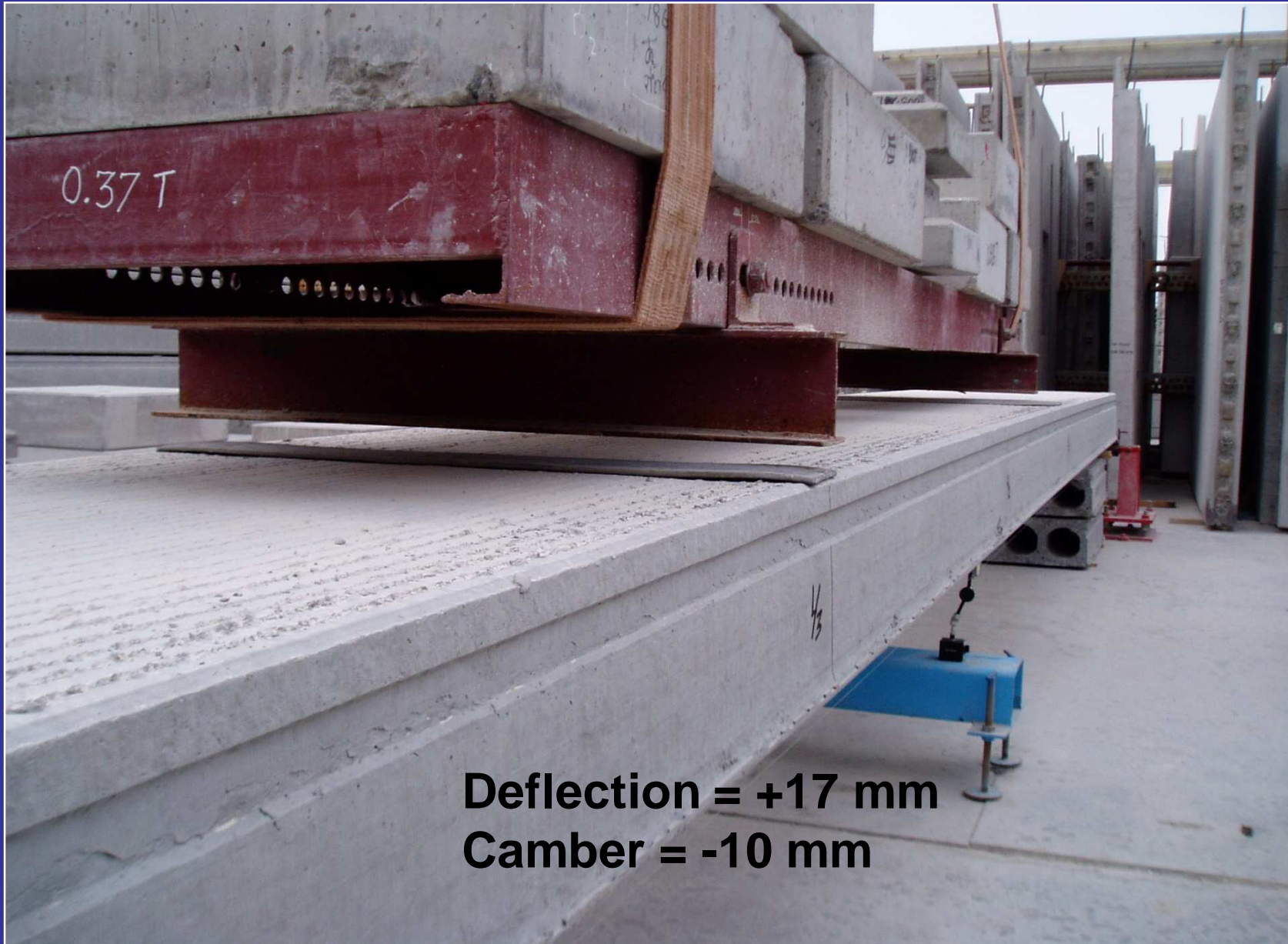


## 4 point bending test of prestressed hollow core slab.

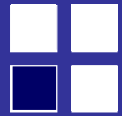




**Self + imposed  $M_{Ed}$  = service moment of resistance**



**Deflection = +17 mm  
Camber = -10 mm**

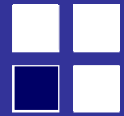


**Self + imposed  $M_{Ed} = M_{Rd}$  (ultimate resistance)**



**First cracking**

**Deflection approx 25 mm**



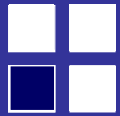
**Self + imposed  $M_{Ed} = 1.25 M_{Rd}$  (ultimate load + 25%)**



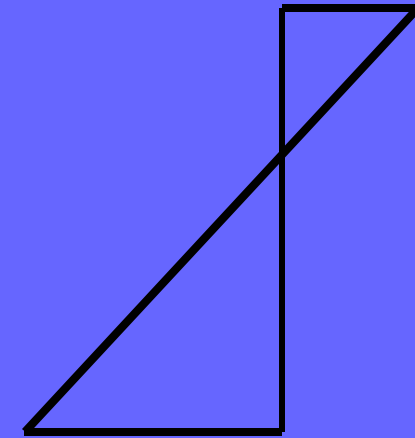
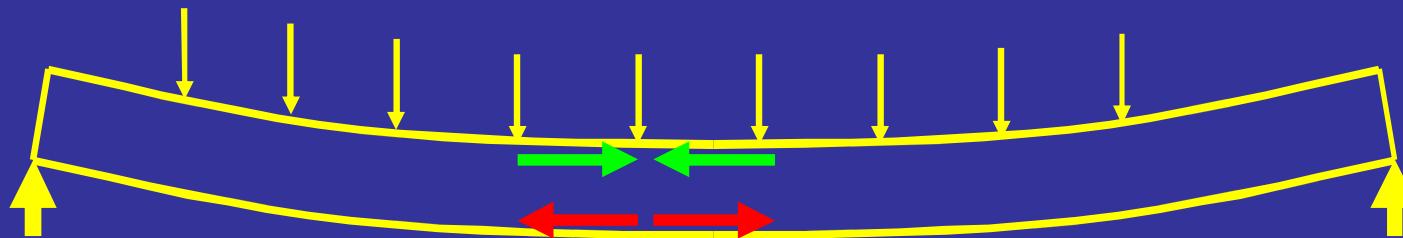
**Cracks widening and increasing**

**Deflection approx 35 mm**

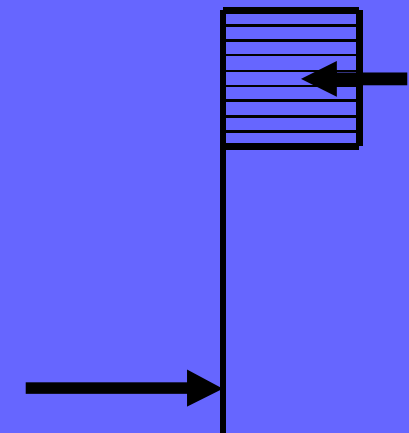
**Limit = span / 250 = 44 mm**



# Ultimate moment of resistance

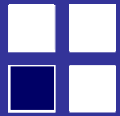


Strains

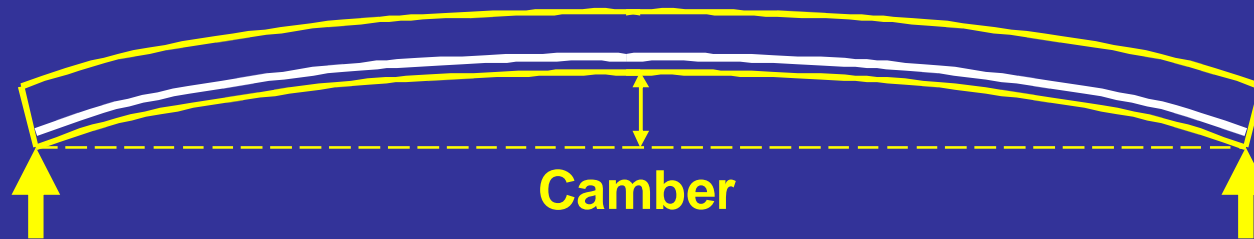


Stress



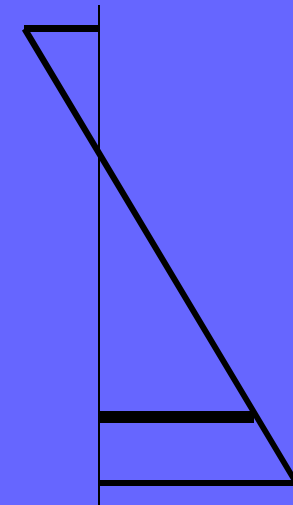


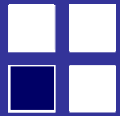
## Strain development from initial prestress to ultimate



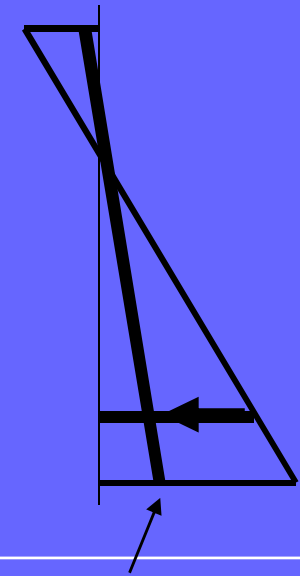
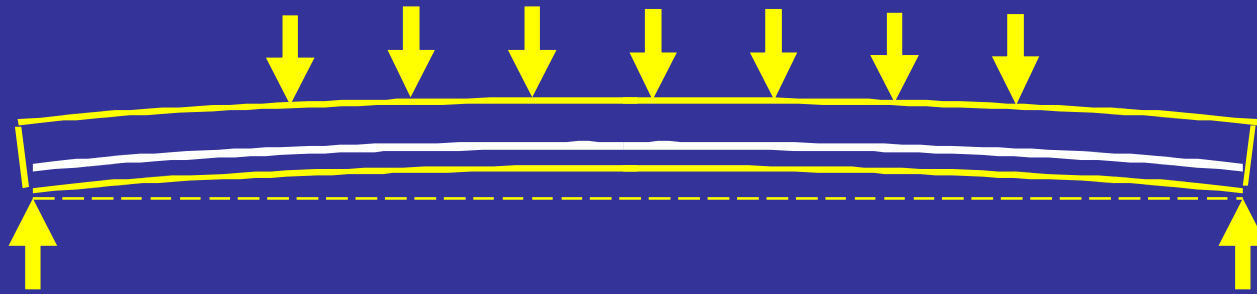
First is the pre-strain due to final prestress after all losses =

$$\epsilon_{po} = \sigma_{po} / E_p$$

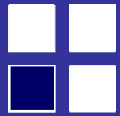




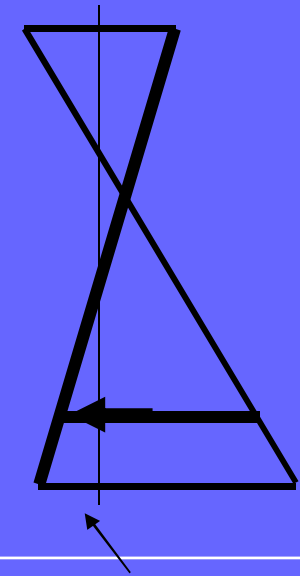
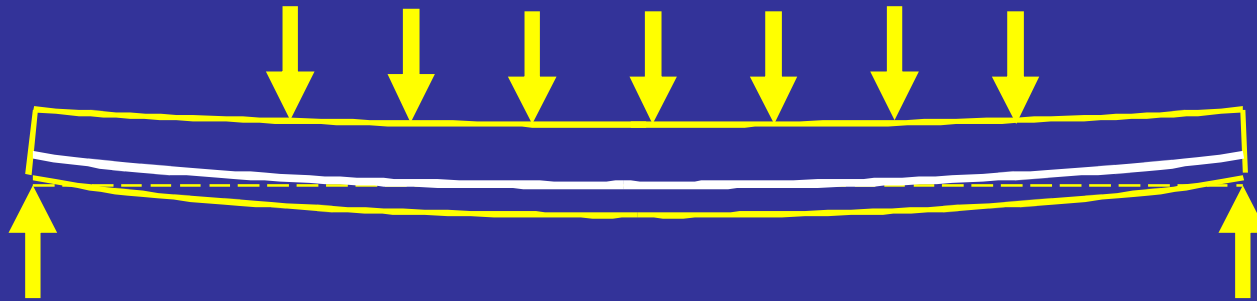
# Strain development from initial prestress to ultimate



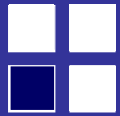
**Bending strain added to pre-strain**



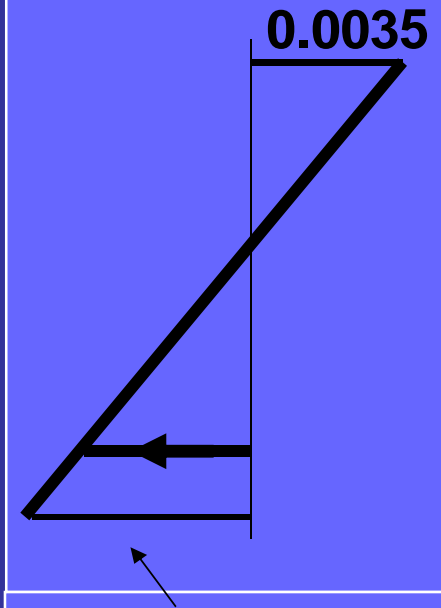
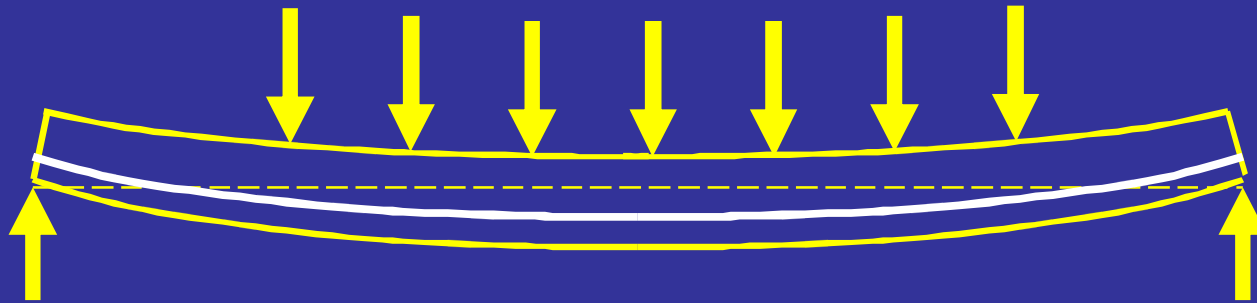
## Strain development from initial prestress to ultimate



**Bending strain  
now overtake  
the pre-strain**



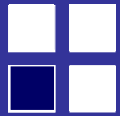
## Strain development from initial prestress to ultimate



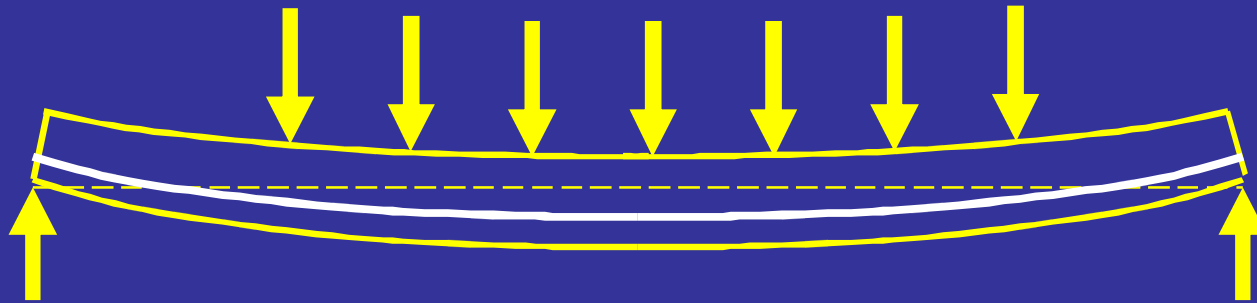
**Final ultimate strains**

$$\epsilon_{cu} = 0.0035$$

$\epsilon_p < \epsilon_{ud}$  **code value 0.02**



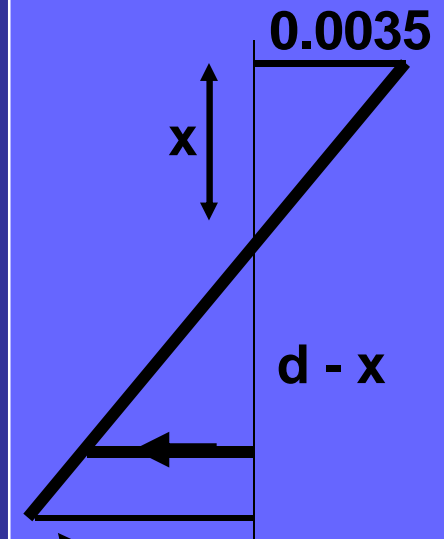
## Strain development from initial prestress to ultimate



### Total strain

$$\epsilon_p = \epsilon_{p0} + 0.0035 (d - x) / x$$

Now find  $x$  and the stresses



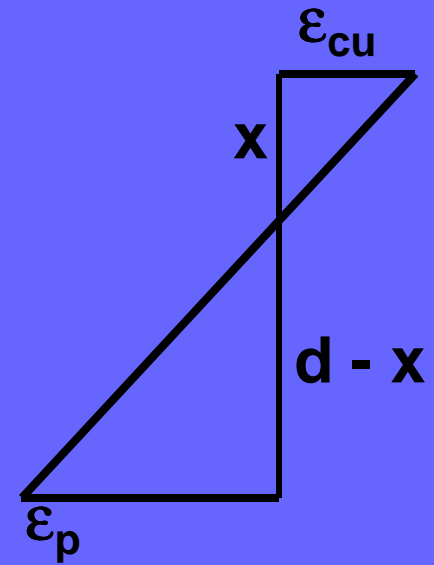
### Final ultimate strains

$$\epsilon_{cu} = 0.0035$$

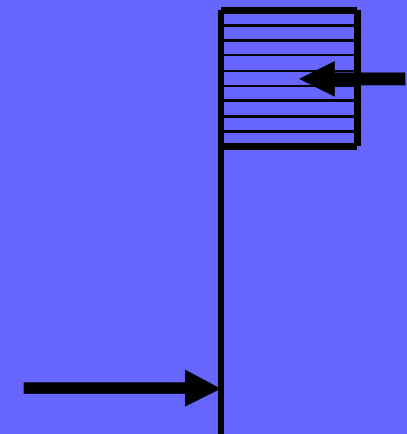
$\epsilon_p < \epsilon_{ud}$  code  
value 0.02



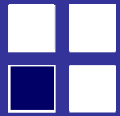
## Constitutive relationship stress v strain



Strains



Stress



## Constitutive relationship stress v strain

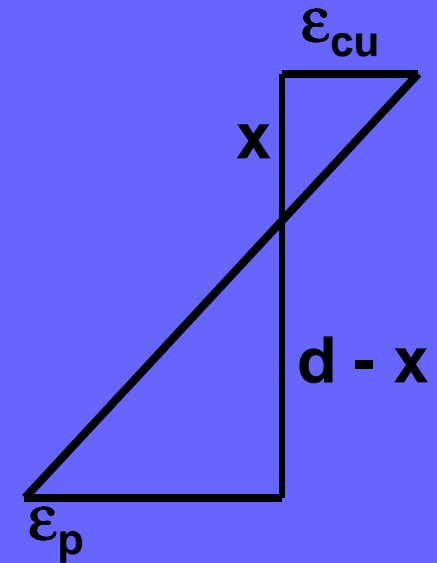
### Equilibrium

$$F_c = F_s$$

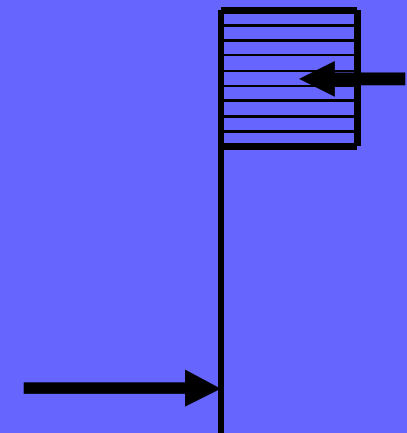
$$0.567 f_{ck} b 0.8 x = f_p A_p$$

### and compatibility

$$\frac{x}{(d - x)} = \frac{\epsilon_{cu}}{\epsilon_p}$$



Strains



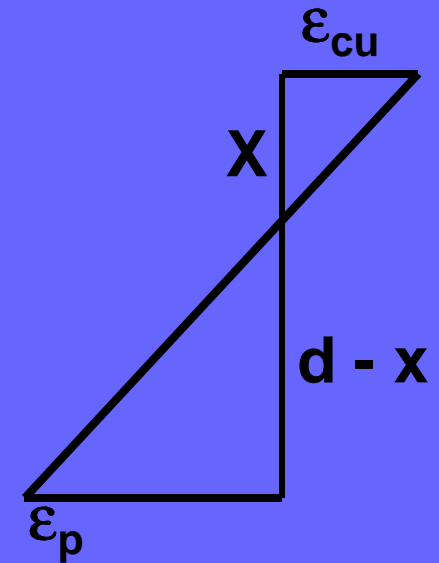
Stress



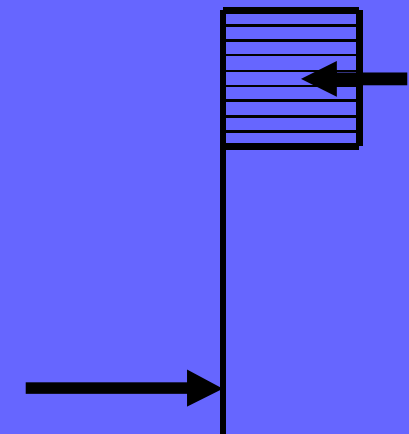
**Combining ;**

$$f_p = 0.567 f_{ck} b 0.8 (d-x) \varepsilon_{cu} / A_p \varepsilon_p$$

**or stress = inverse of strain**

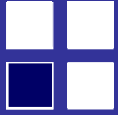


**Strains**



**Stress**

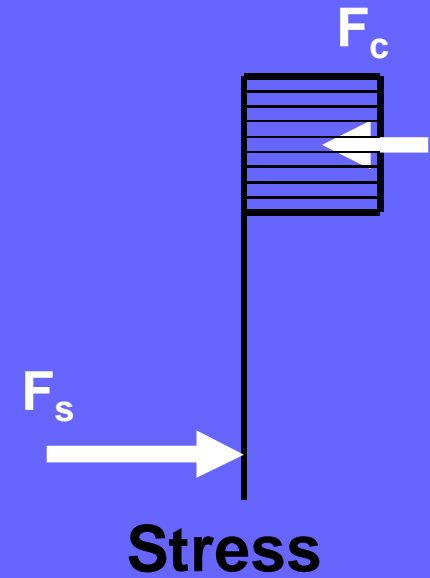
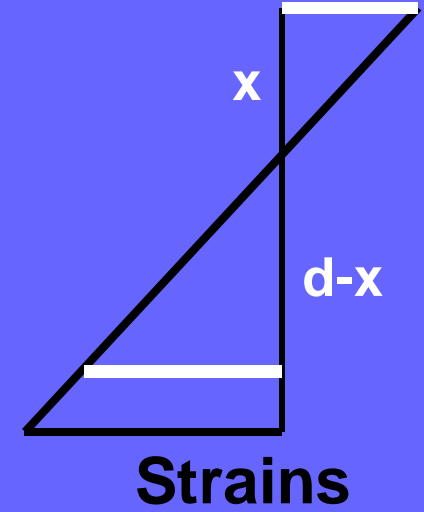




Stress  $f_p$

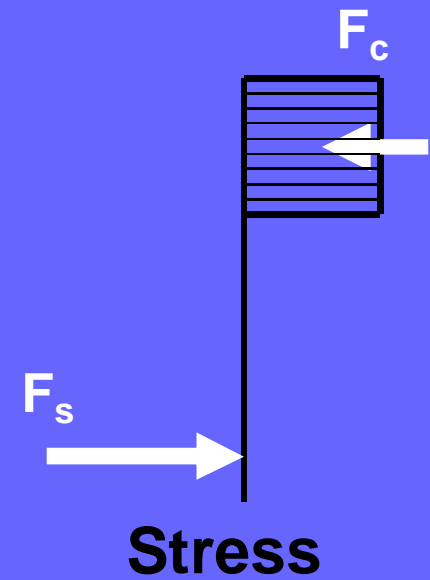
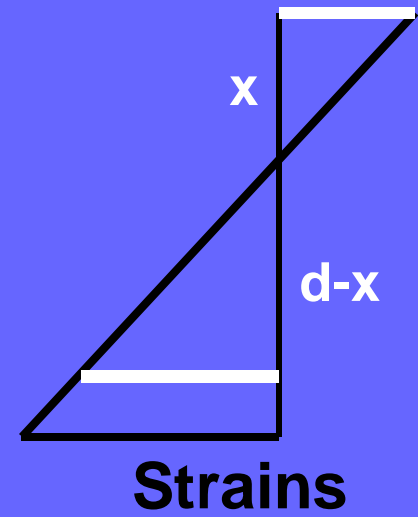
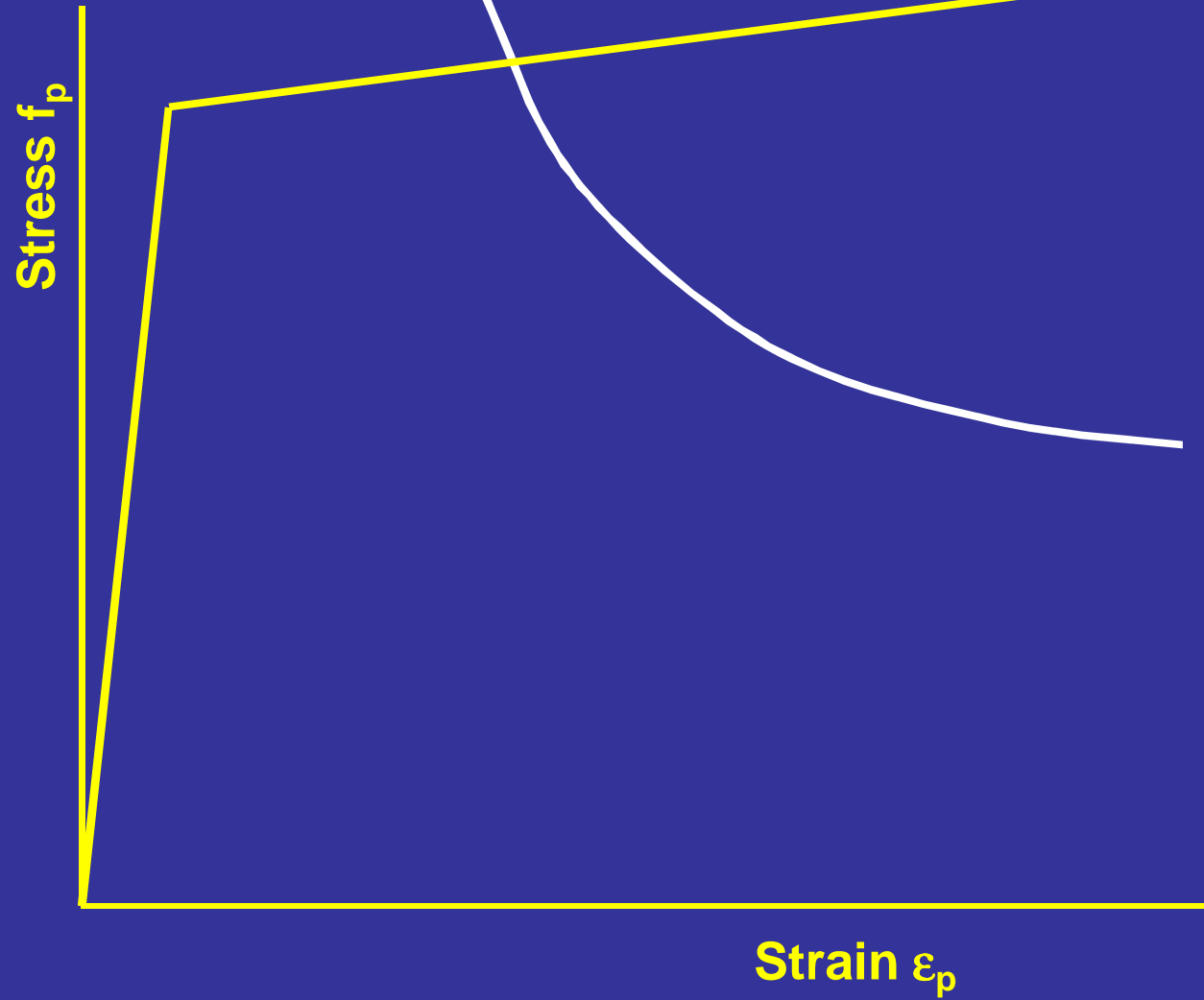
1. Equilibrium of forces gives inverse stress v strain relationship

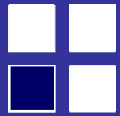
Strain  $\epsilon_p$



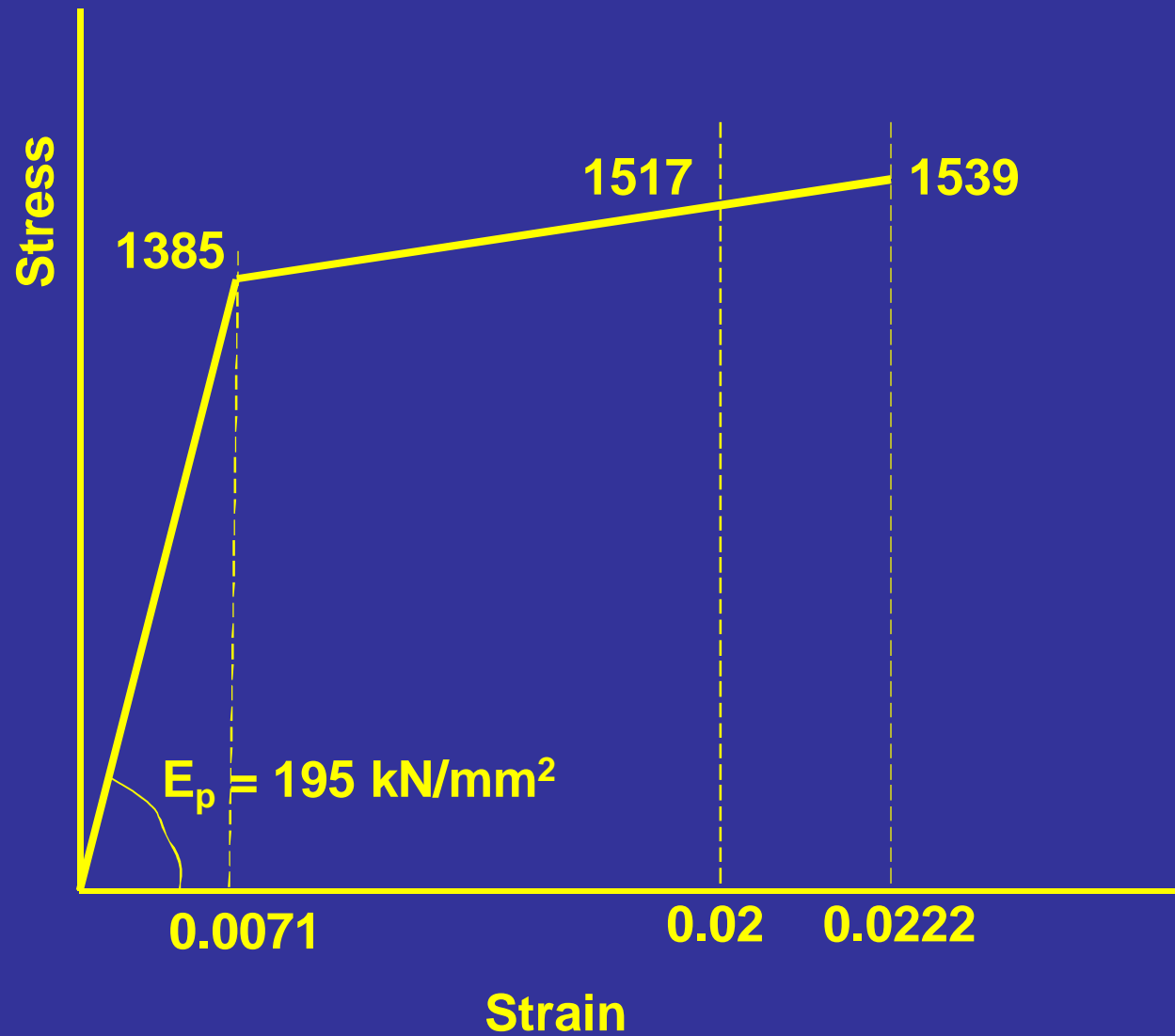


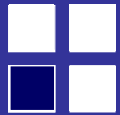
## 2. Stress v strain curve for strand



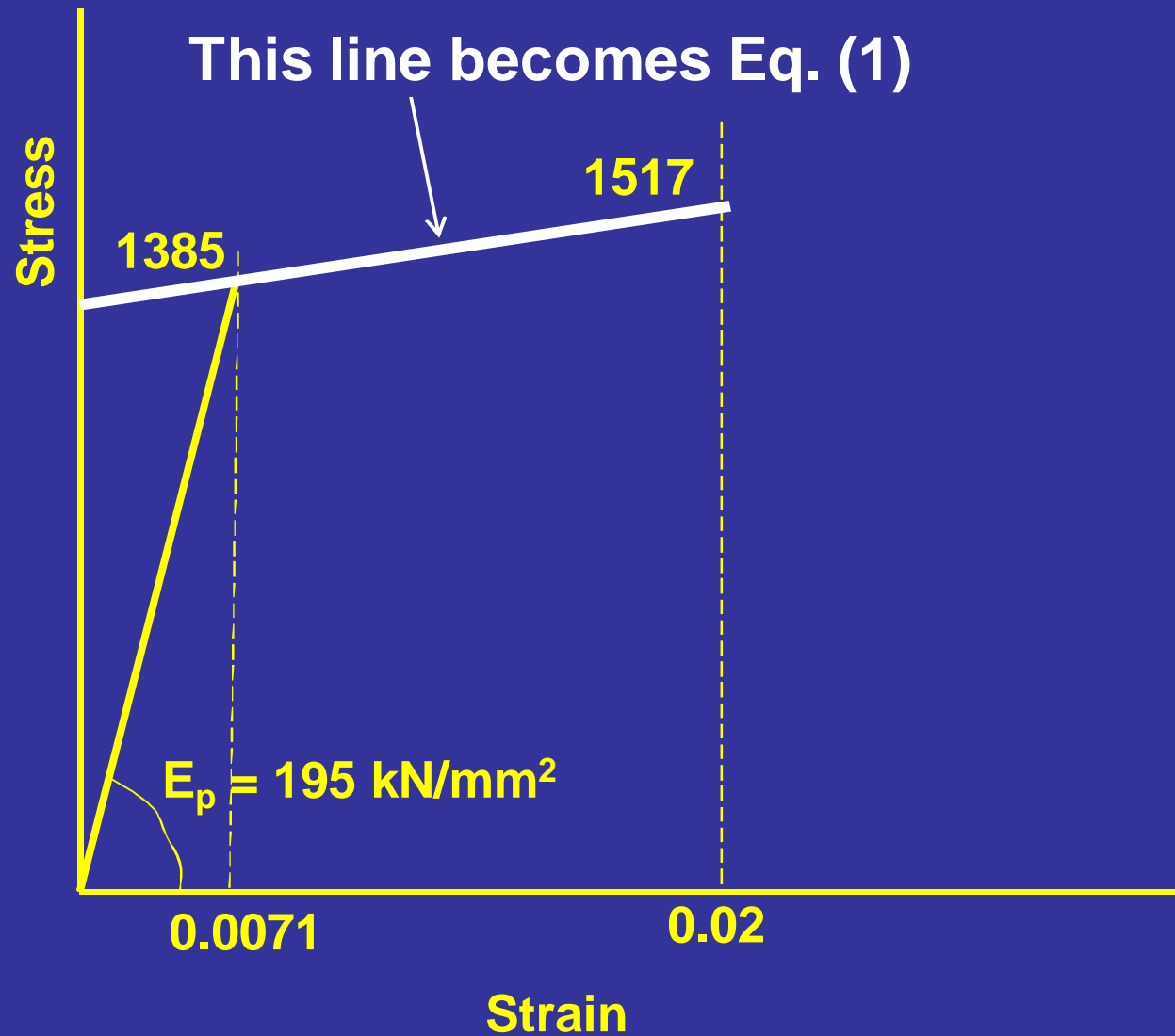


# Stress v strain diagram, e.g. for strand with $f_{pk} = 1770 \text{ N/mm}^2$





# Stress v strain diagram, e.g. for strand with $f_{pk} = 1770 \text{ N/mm}^2$





**Total strain = pre-strain + compatibility**  
**concrete strain  $\varepsilon_p$**

$$= \varepsilon_{po} + \varepsilon_{cu} (d / x - 1) \quad \dots(2)$$

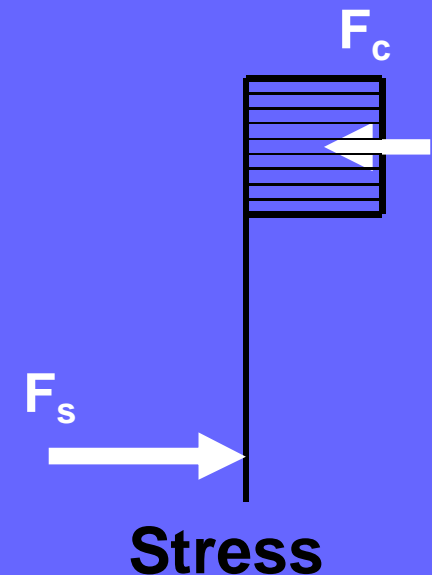
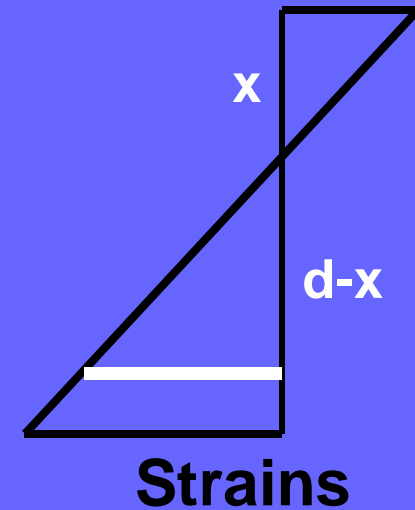
**where, pre-strain after losses**

$$\varepsilon_{po} = \sigma_{po} / E_p$$

**Force equilibrium**

$$F_s = F_c$$

$$f_p A_p = 0.567 f_{ck} b 0.8 x \quad \dots(3)$$





**Combining 3 equations gives the quadratic solution:**

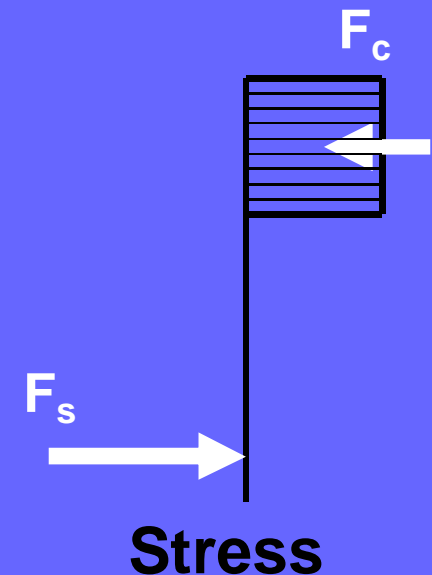
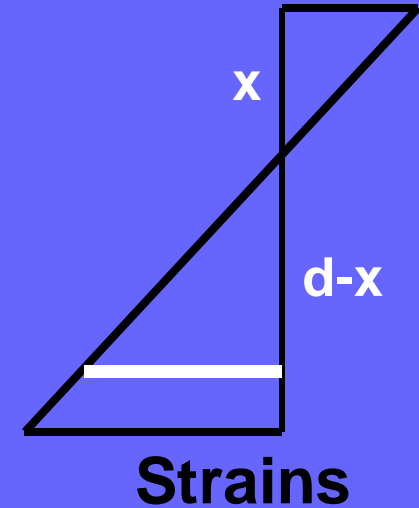
$$0.567 f_{ck} 0.8 b (\varepsilon_{uk} - \varepsilon_{LOP}) x^2$$

$$-[0.9(\varepsilon_{uk} - \varepsilon_{LOP}) + 0.1(\varepsilon_{po} - \varepsilon_{cu} - \varepsilon_{LOP})] A_p f_{pd} x$$

$$-0.1\varepsilon_{cu} d A_p f_{pd} = 0$$

**Solving yields x**

**Then  $\varepsilon_p$  and  $f_p$  are found**



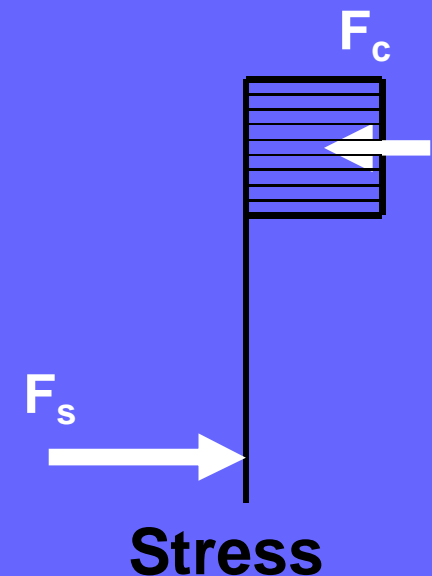
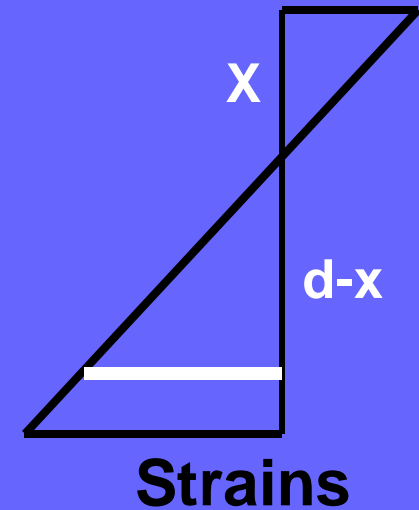


Check that  $f_p$  is not greater than the maximum allowed, e.g.  $f_{pk,max} = 1517 \text{ N/mm}^2$

Check  $0.8x < \text{depth of top flange}$

Determine the centroid of the compression block,  $d_n = 0.4x$

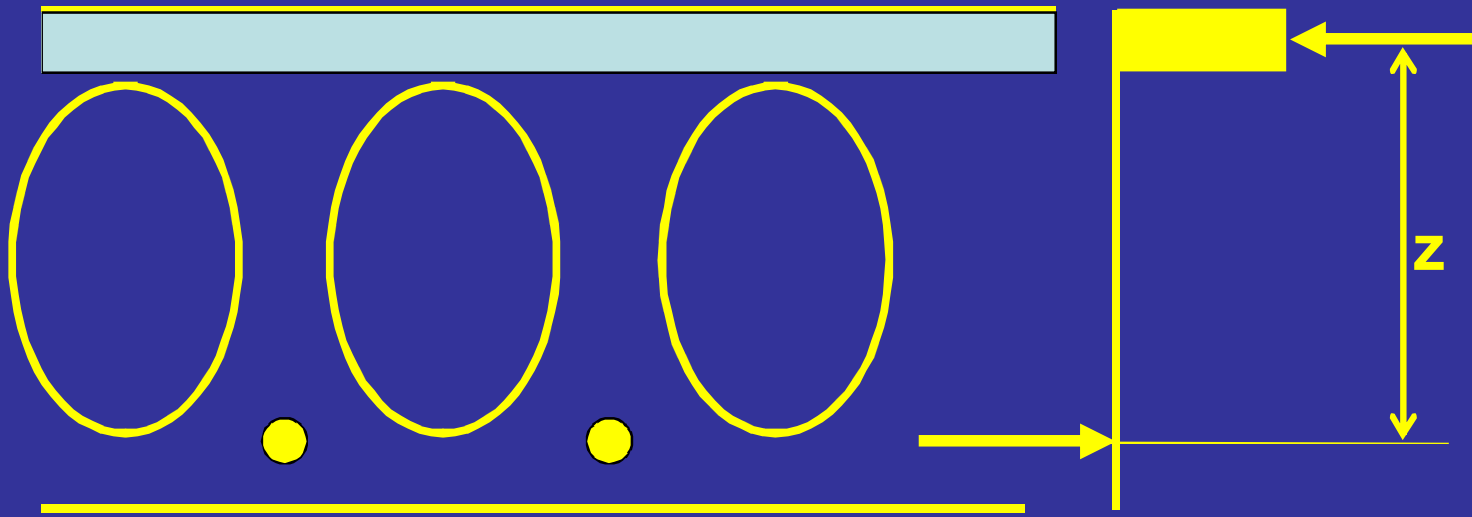
Lever arm  $z = d - d_n$



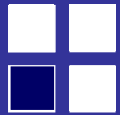


## Ultimate moment of resistance

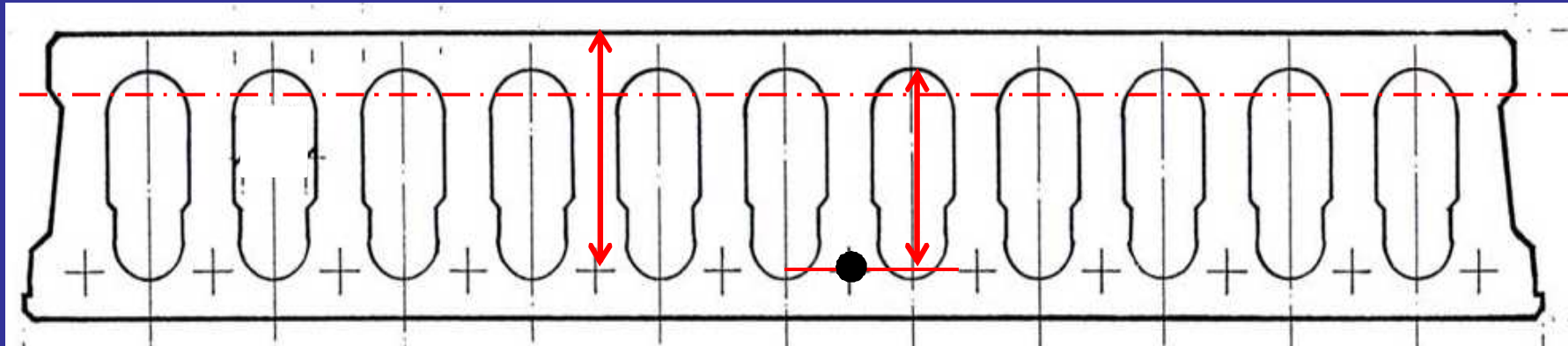
$$M_{Rd} = f_p A_p z$$







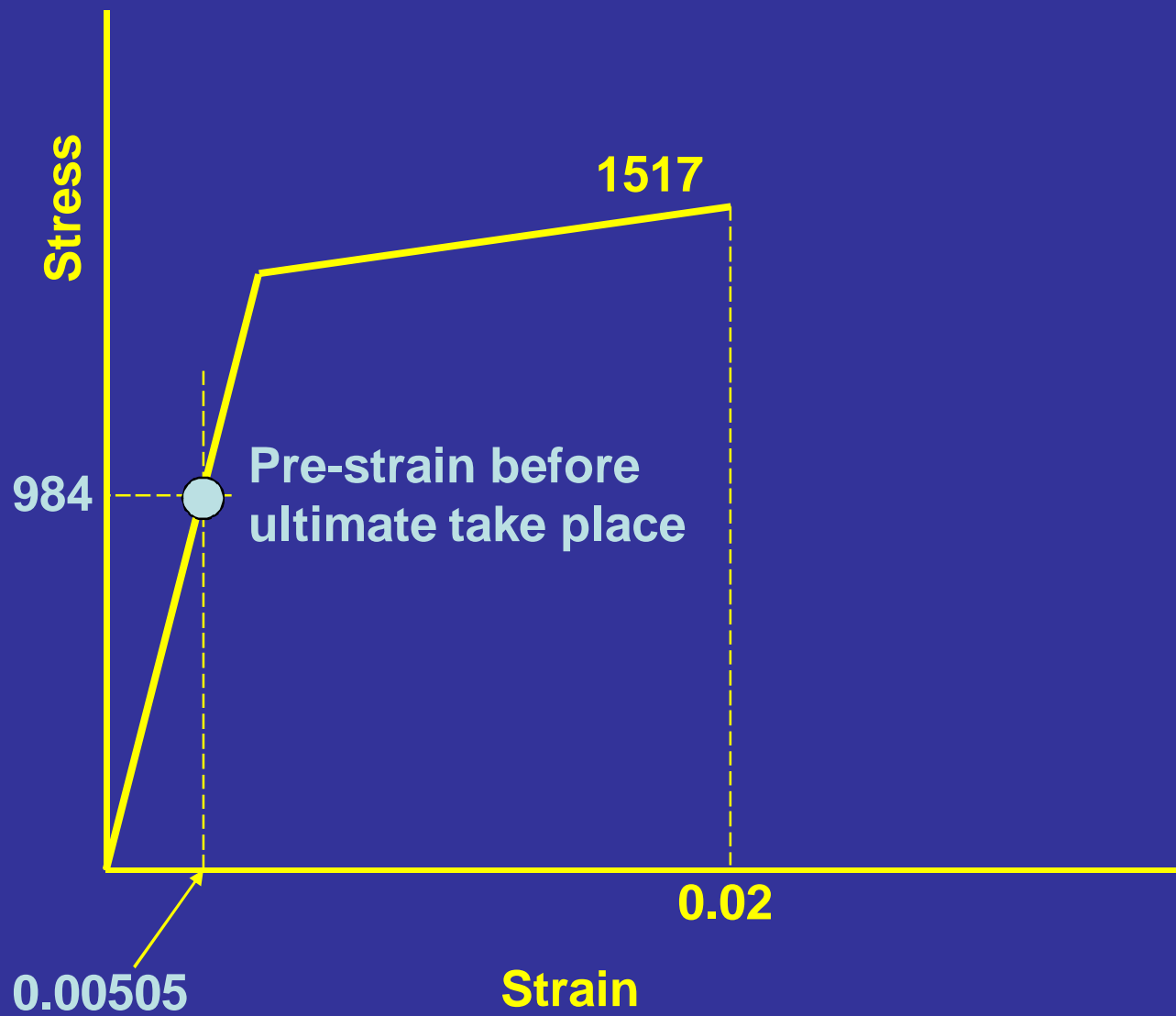
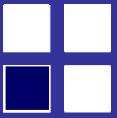
## Worked example (continued)

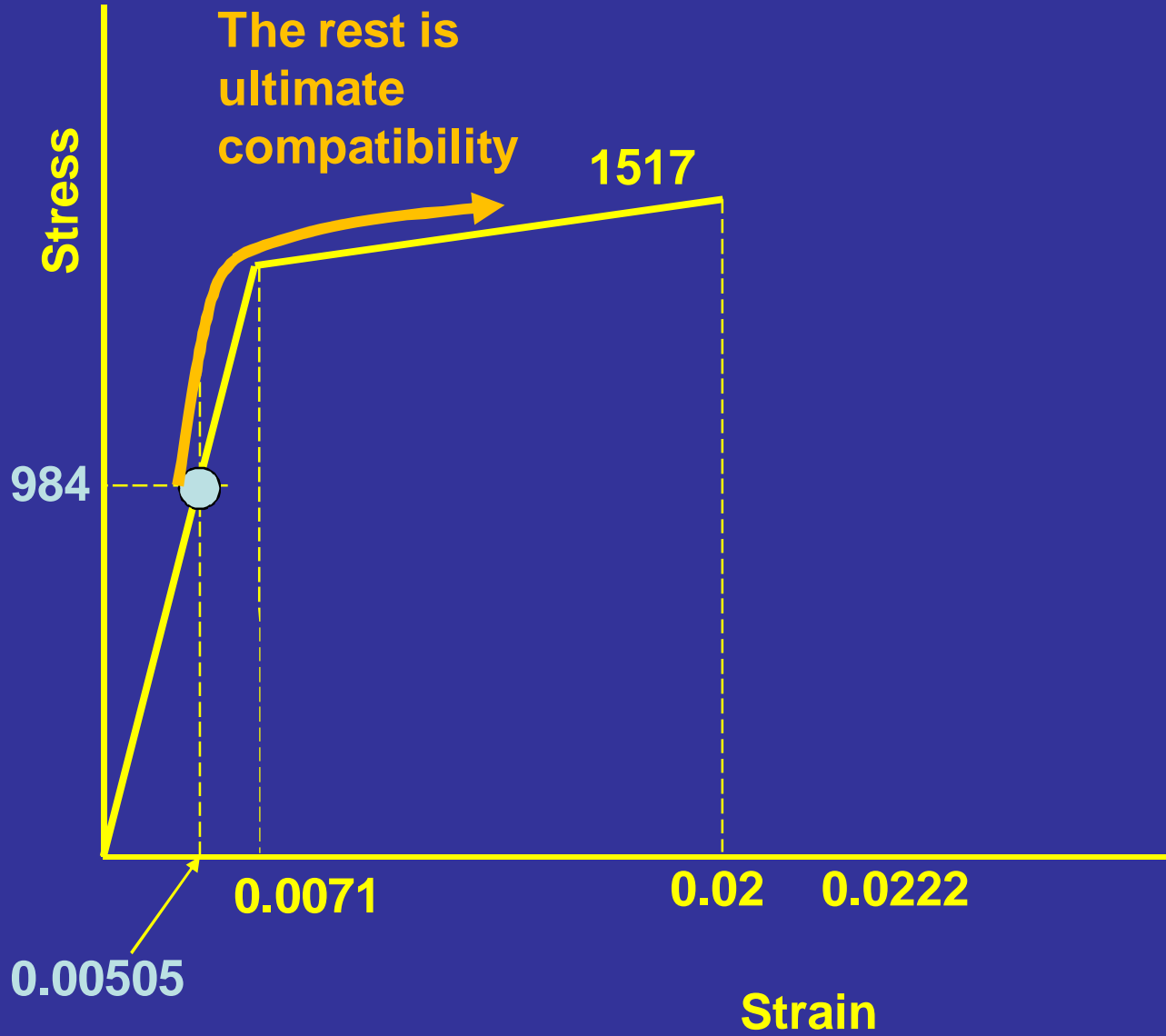
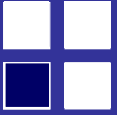


$$A_p = 624 \text{ mm}^2$$

$$d = 250 - 39.6 = 210.4 \text{ mm}$$

$$\varepsilon_{po} = 984.5 / 195000 = 0.00505$$







## Worked example

The quadratic terms are:

$$369.6 x^2 - 12515 x - 70708 = 0$$

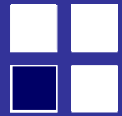
$$x = 38.8 \text{ mm}$$

$$\text{Compression depth} = 0.8 \times 38.8 = 31.0 \text{ mm}$$

$$< \text{top flange depth} = 35 \text{ mm}$$

$$d_n = 0.4 \times 38.8 = 15.5 \text{ mm}$$

$$z = 210.4 - 15.5 = 194.9 \text{ mm}$$



## Worked example

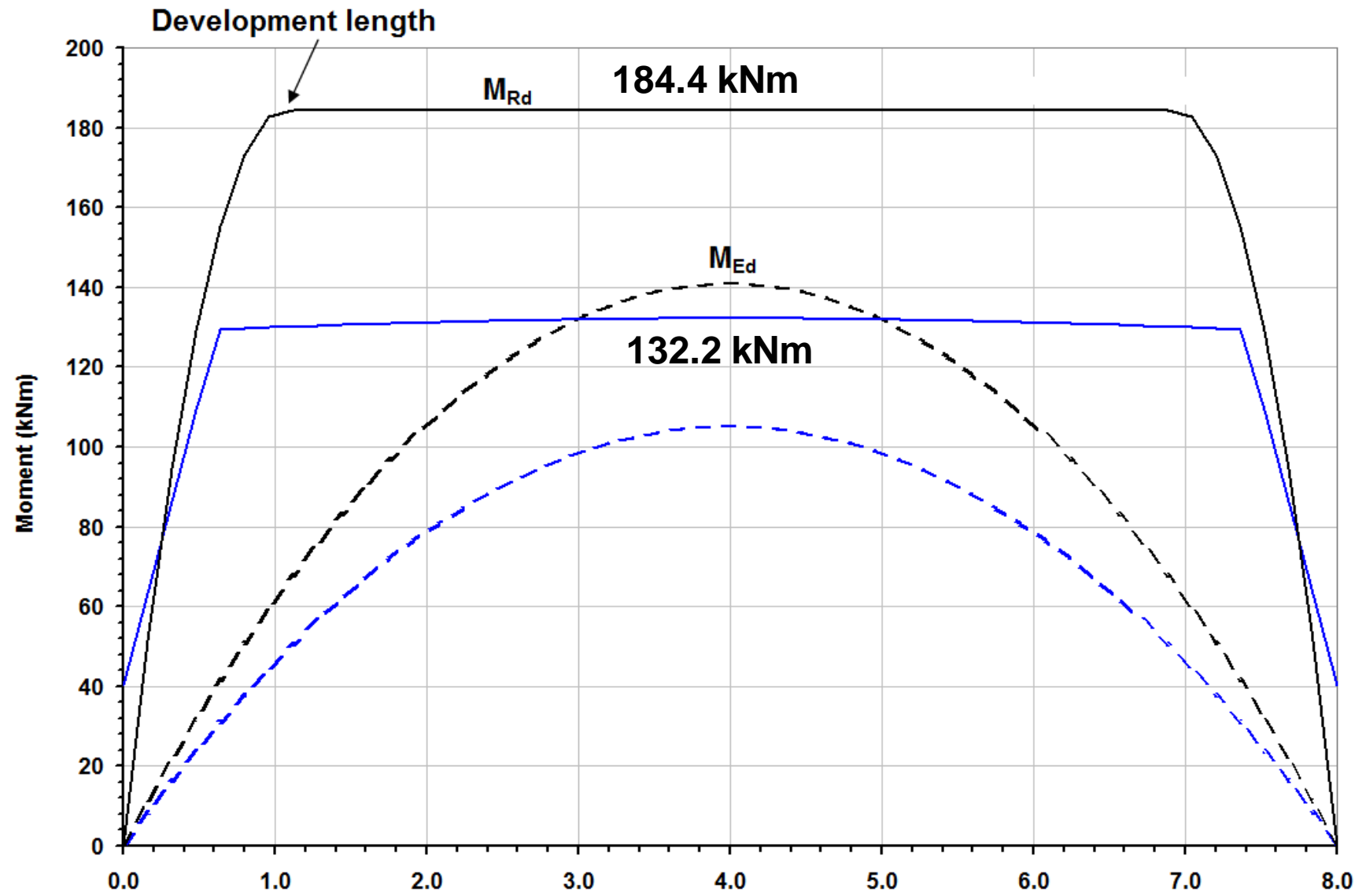
Then  $\varepsilon_p = 0.202025 > 0.02$

$\therefore f_p = 1517 \text{ N/mm}^2$

$M_{Rd} = 624 \times 1517 \times 194.9 \times 10^{-6} = \underline{184.4 \text{ kNm}}$

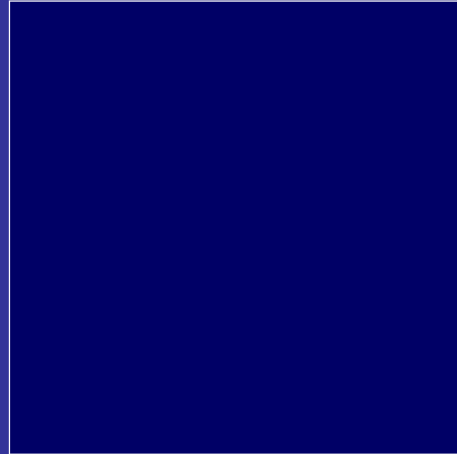
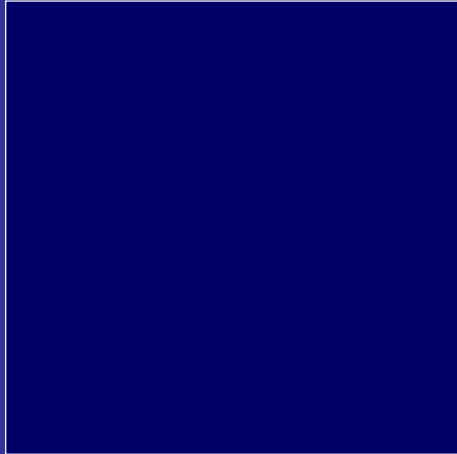
Remember  $M_{sd} = 132.2 \text{ kNm} \quad \therefore M_{Rd} / M_s = 1.39$

**A good margin for most dead and live load combinations**



# Syllabus

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**Camber.**  
**Creep.**  
**Deflections.**  
**Limits.**





**Pre-camber,  
here 3 days  
after transfer**







## Camber & Deflections

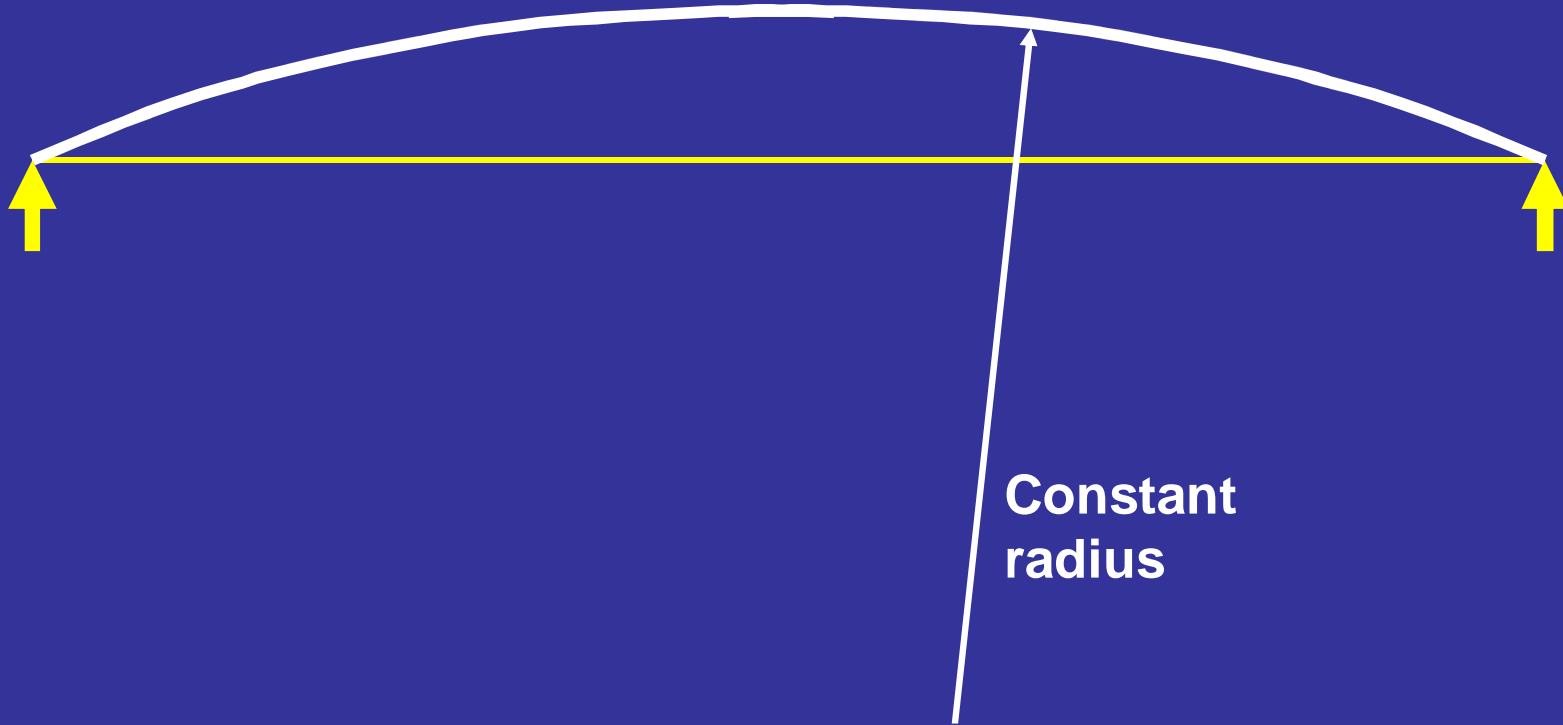
1. Pre-camber at transfer  $< L/300 \pm 50\%$
2. Deflection due to self weight at installation  $< L/250$
3. Long-term total deflection  $< L/250^*$
4. Active deflection (after installation)  $< L/500^*$  (or  $L/350$  if non-brittle finishes)

\* EC2-1-1 limits



## Upward camber due to transfer force

$$\delta_1 = - P_{pm0} z_{cp} L^2 / 8 E_{cm}(t) I_{xx}$$



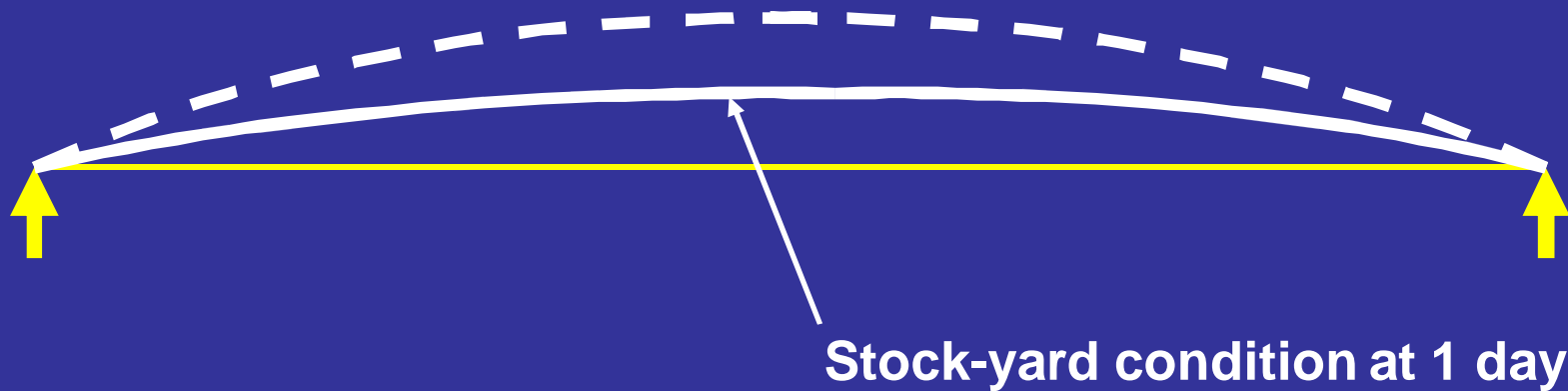


## Upward camber due to transfer force

$$\delta_1 = - P_{pm0} z_{cp} L^2 / 8 E_{cm}(t) I_{xx}$$

plus downward due to self weight

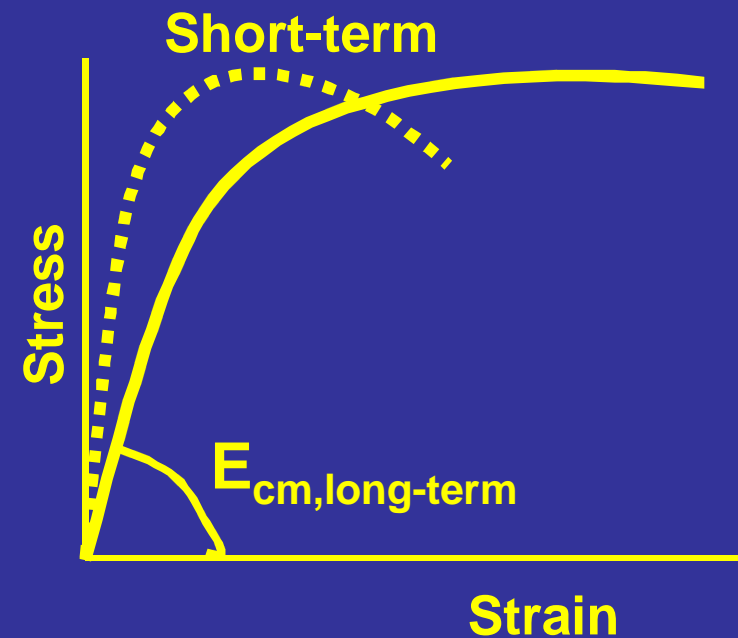
$$\delta_2 = +5 w_o L^4 / 384 E_{cm}(t) I_{xx}$$





## Camber & Deflections

Creep of concrete causes a reduction in Young's modulus, but at the same time the concrete is gaining strength and stiffness to 28 days.





## Camber & Deflections

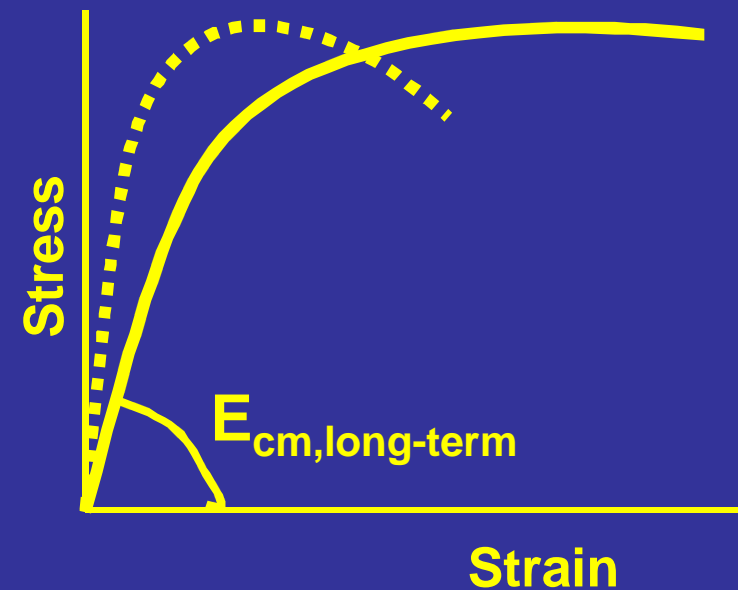
Creep of concrete causes a reduction in Young's modulus, but at the same time the concrete is gaining strength and stiffness to 28 days.

Creep coefficient  $\varphi_{\infty} = 2.5$

Coefficient of development at:

transfer	= 0.1
15 days	= 0.3
28 days	= 0.4
2 months	= 0.5
3 months	= 0.6
$\infty$	= 1.0

Values from ASSAP, Italy





## Camber & Deflections

Creep coefficient  $\varphi_{\infty} = 2.5$

Coefficient of development at:

transfer = 0.1

28 days = 0.4

So the net effect is to average the 1 and 28 day values

$$\varphi_1 = E_{cm}(t) / 0.5 \times [E_{cm} + E_{cm}(t)]$$



## Camber & Deflections

Creep coefficient  $\varphi_{\infty} = 2.5$

Coefficient of development at:

transfer = 0.1

28 days = 0.4

$$\varphi_1 = E_{cm}(t) / 0.5 \times [E_{cm} + E_{cm}(t)] \times 2.5 \times (0.4 - 0.1)$$

$$= 0.75 \times E_{cm}(t) / 0.5 \times [E_{cm} + E_{cm}(t)]$$

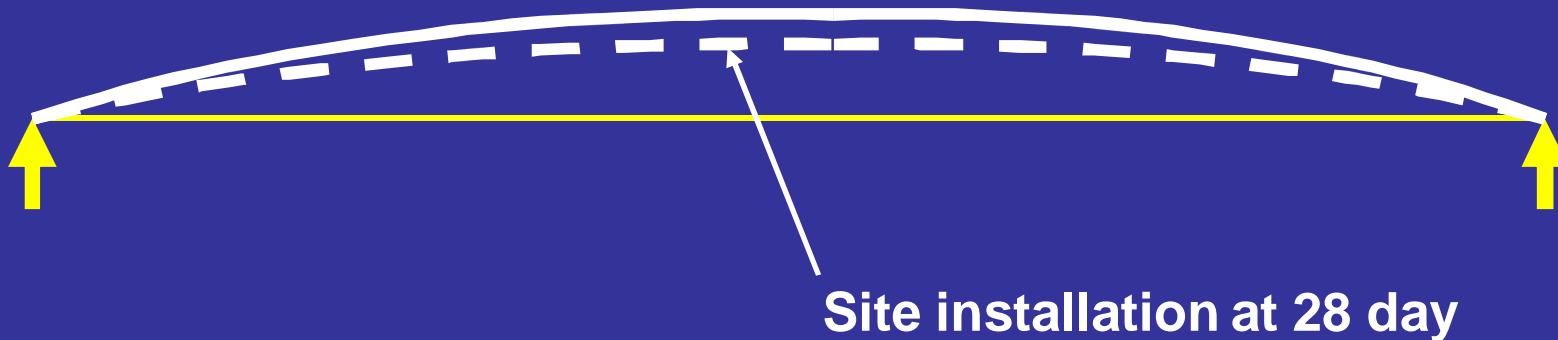


At 28 days, - creep camber + a bit for the small change in prestress force + creep deflection =

$$\delta_3 = - (1 + \varphi_1) \delta_1 + (P_{pm0} - P_{pmi}) z_{cp} L^2 / 8 E_{cm} I_{xx}$$

plus downward due to self weight

$$\delta_4 = + (1 + \varphi_1) \delta_2$$



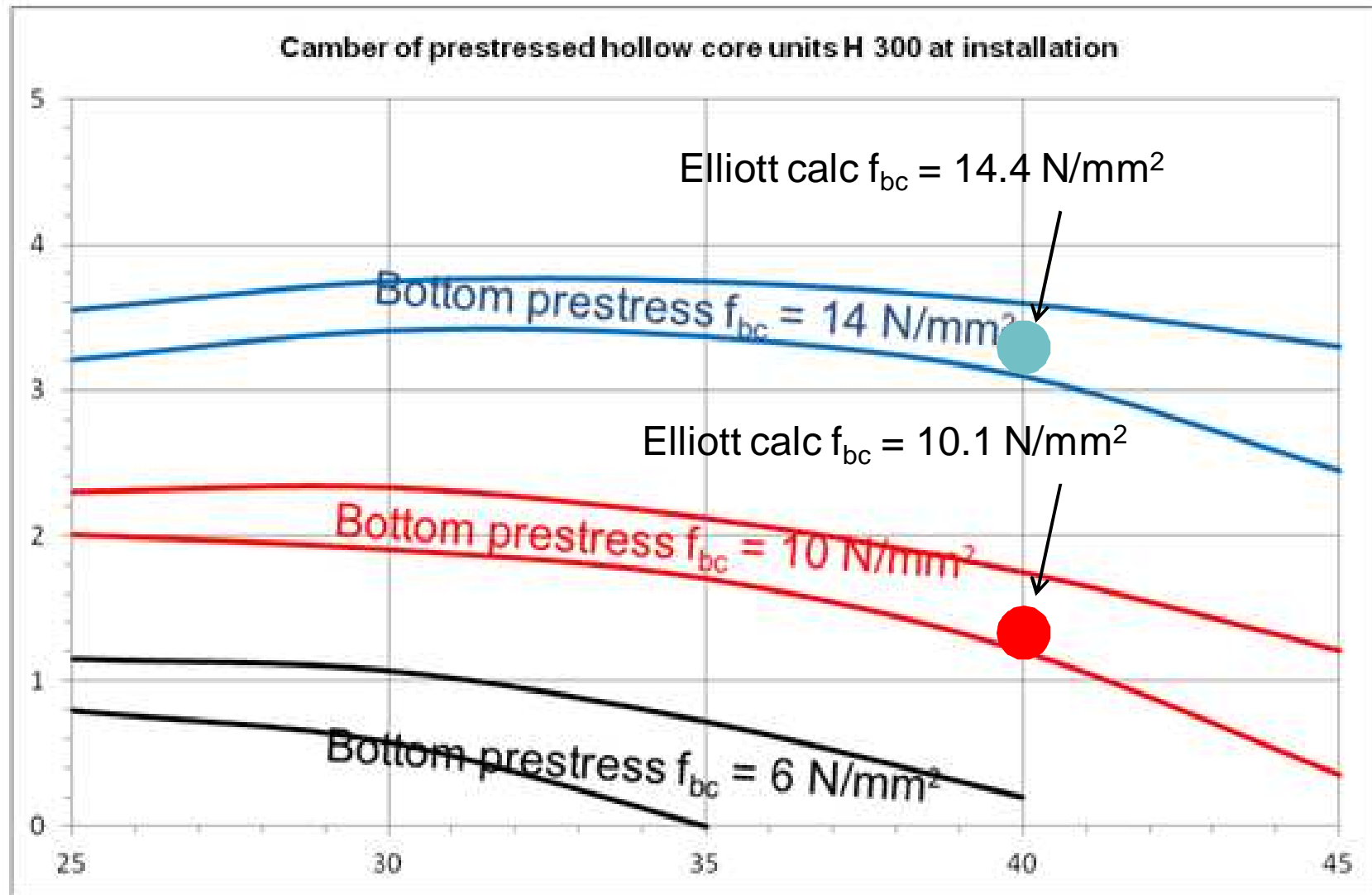




# fib Manual

## Camber at installation for 300 mm deep hcu

Normalised upward camber = 1000 v/L





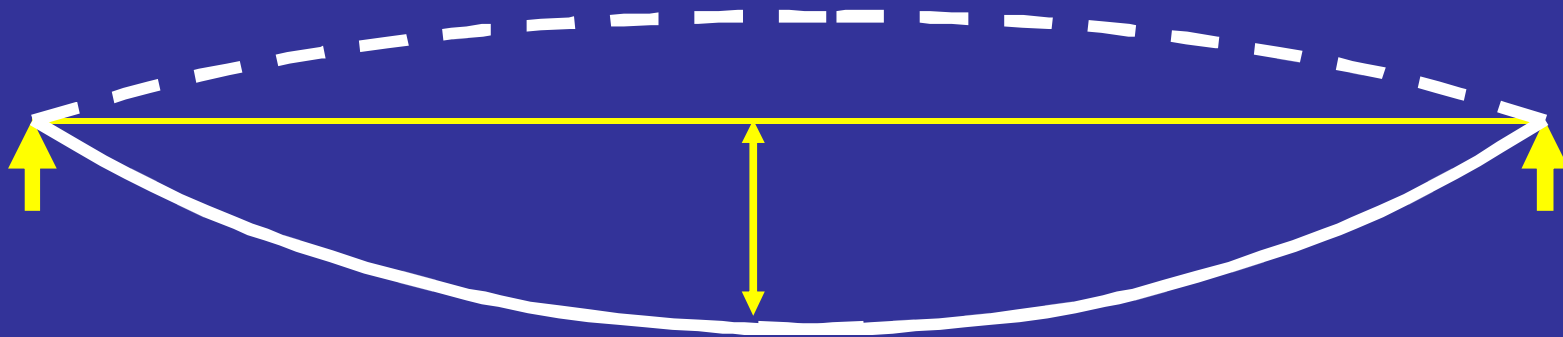
## Long-term changes from $E_{cm}$ to $E_{cm} / (1 + \varphi_{\infty})$

$$0.8\varphi_{\infty} = 0.8 \times 2.5 = 2.0$$

0.8 is a long-term concrete aging coefficient

For loads after installation

$$\varphi_{28} = 2.0 \times (1.0 - 0.4) = 1.20$$

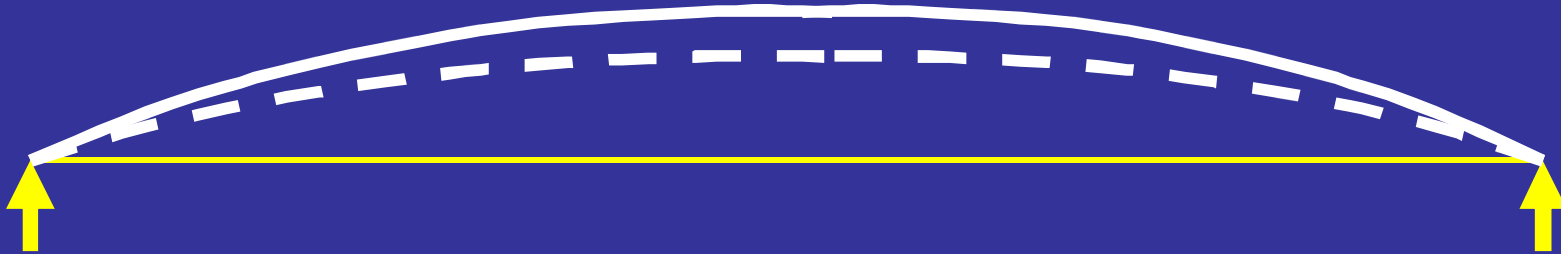


**Final long-term deflection  
from many sources**



First, camber increases upwards, less a bit for the change in prestress

$$\delta_5 = -\delta_3 + [\phi_{28} P_{pmi} - (P_{pmi} - P_{po})] z_{cp} L^2 / 8 E_{cm} I_{xx}$$

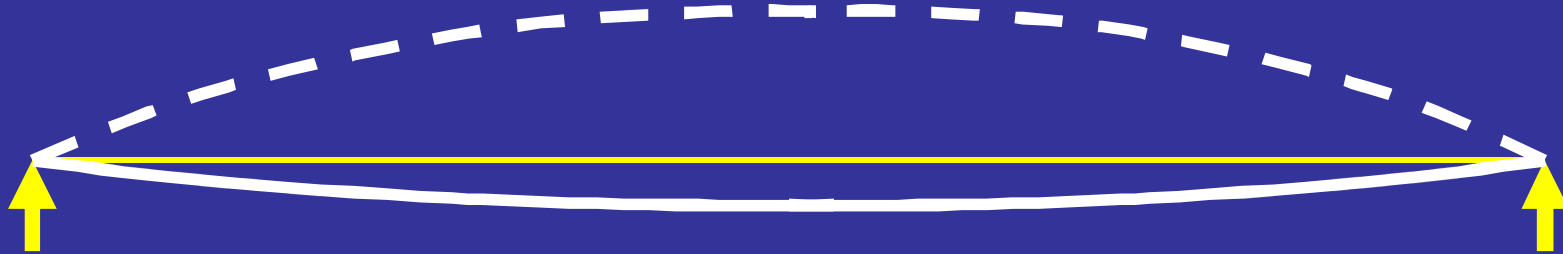




..then self weight creeps down, 2<sup>nd</sup> term is the creep

$$\delta_5 = -\delta_3 + [\varphi_{28} P_{pmi} - (P_{pmi} - P_{po})] z_{cp} L^2 / 8 E_{cm} I_{xx}$$

$$\delta_6 = +\delta_4 + 5 w_1 \varphi_{28} L^4 / 384 E_{cm} I_{xx}$$





**..followed by finishes, dead loads  $w_2$  after 28 days**

$$\delta_5 = -\delta_3 + [\varphi_{28} P_{pmi} - (P_{pmi} - P_{po})] z_{cp} L^2 / 8 E_{cm} I_{xx}$$

$$\delta_6 = +\delta_4 + 5 w_1 \varphi_{28} L^4 / 384 E_{cm} I_{xx}$$

$$\delta_7 = + (1 + \varphi_{28}) 5 w_2 L^4 / 384 E_{cm} I_{xx}$$





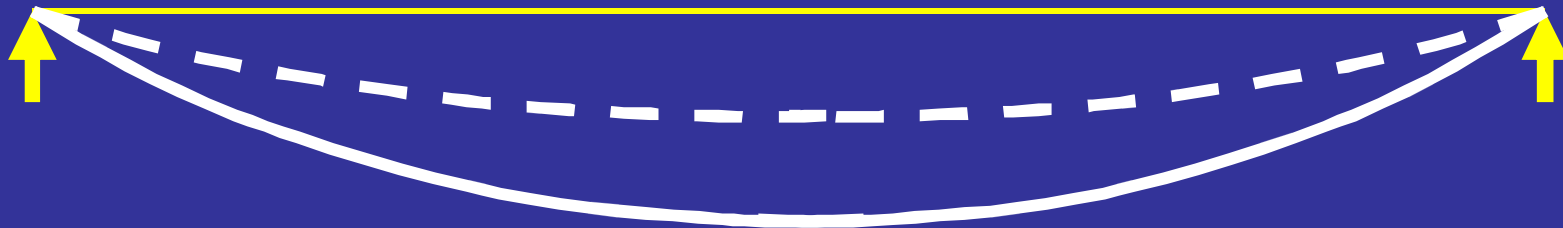
**..and finally live loads  $\psi_2 w_3$  over infinity time**

$$\delta_5 = -\delta_3 + [\varphi_{28} P_{pmi} - (P_{pmi} - P_{po})] z_{cp} L^2 / 8 E_{cm} I_{xx}$$

$$\delta_6 = +\delta_4 + 5 w_1 \varphi_{28} L^4 / 384 E_{cm} I_{xx}$$

$$\delta_7 = + (1 + \varphi_{28}) 5 w_2 L^4 / 384 E_{cm} I_{xx}$$

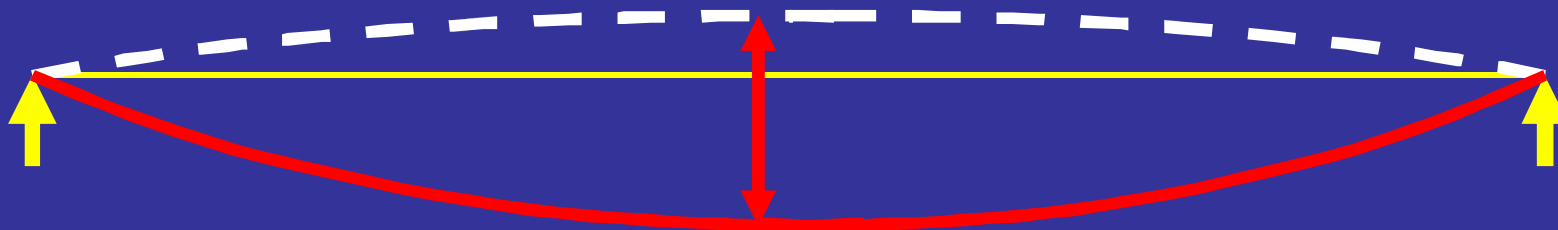
$$\delta_8 = + (1 + 0.8 \varphi_{\infty}) 5 \psi_2 w_3 L^4 / 384 E_{cm} I_{xx}$$





## Active deflections due to creep effects and live loads takes parts of the previous equations

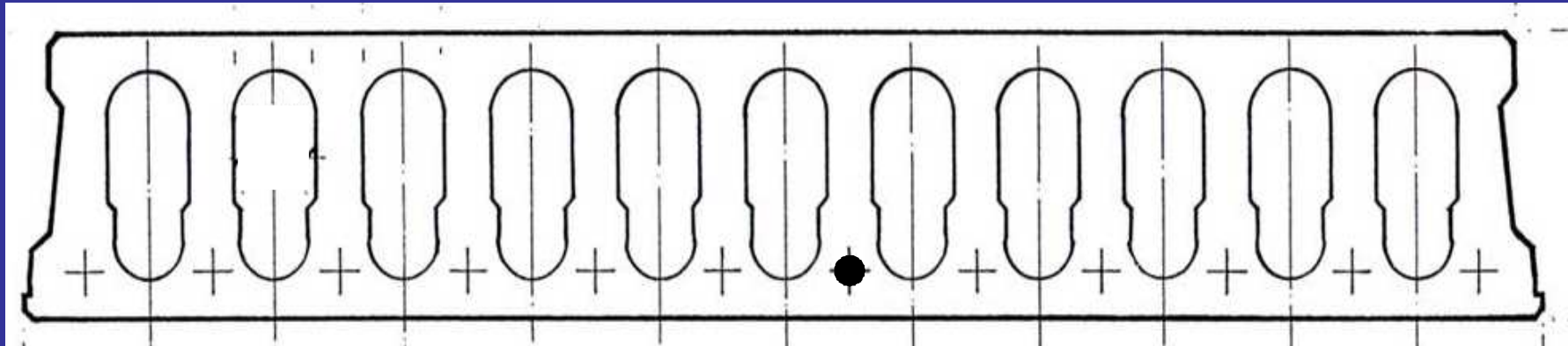
$$\begin{aligned}\delta_g = & [\varphi_{28} P_{pmi} - (P_{pmi} - P_{po})] z_{cp} L^2 / 8 E_{cm} I_{xx} \\ & + \varphi_{28} 5 (w_1 + w_2) L^4 / 384 E_{cm} I_{xx} \\ & + (1 + 0.8 \varphi_{\infty}) 5 \psi_2 w_3 L^4 / 384 E_{cm} I_{xx}\end{aligned}$$



For composite design, replace  $I_{xx}$  with  $I_{xx,c}$



## Worked example (continued)



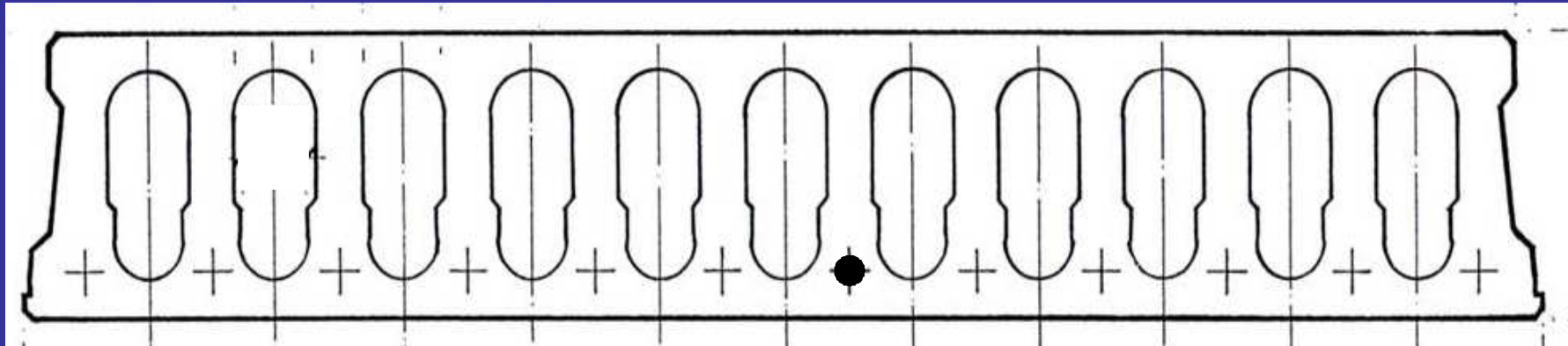
Calculate camber, installation and long-term deflection

8.0 m effective span





## Worked example



**Self weight =  $182791 \times 24.5 \times 10^{-6} = \underline{4.48 \text{ kN/m}}$**

**Dead loads =  $3.0 \text{ kN/m}^2 = \underline{3.60 \text{ kN/m}}$  per unit**

**Use of floor = offices, then  $\psi_2 = 0.3$**

**Live load =  $0.3 \times 4.0 = 1.2 \text{ kN/m}^2 = \underline{1.44 \text{ kN/m}}$  per unit**



## Worked example

### Camber at transfer

$$\delta_1 = \frac{739.0 \times 10^3 \times 82.75 \times 8000^2}{8 \times 32837 \times 1289 \times 10^6} = - 11.6 \text{ mm}$$

### Self weight

Self weight hcu only


$$\delta_2 = \frac{5 \times 4.48 \times 8000^4}{384 \times 32837 \times 1289 \times 10^6} = + 5.7 \text{ mm}$$

Net camber = - 5.9 mm < length / 300 = 26 mm



## Camber at installation

$$\varphi_1 = 2.5 \times (0.4 - 0.1) \times \frac{32837}{0.5 \times (32837 + 36283)} = 0.71$$

$$\delta_3 = - 11.6 \times (1 + 0.71) = - 19.8 \text{ mm}$$

## Self weight at installation

$$\delta_4 = +5.7 \times (1 + 0.71) = +9.7 \text{ mm}$$



## Long term camber

$$\varphi_{\infty} = 0.8 \times 2.5 = 2.0 \text{ for live load}$$

$$\varphi_{28} = 2.0 \times (1 - 0.4) = 1.2 \text{ for creep of camber and dead load}$$

$$\delta_5 = -19.8 - \frac{[717.8 \times 1.2 - (717.8 - 614.3)] \times 82.75 \times 8000^2}{8 \times 36283 \times 1289 \times 10^6}$$

$$= -19.8 - 10.7 = -30.5 \text{ mm}$$



## Long term dead + live

$$\delta_6 = +9.7 + \frac{\begin{array}{ccc} \text{hcu + infill} & \text{dead} & \text{quasi-live} \\ \swarrow & \swarrow & \swarrow \\ 5 \times (1.2 \times 4.73 + 2.2 \times 3.6 + 3.0 \times 1.44) \times 8000^4 & & \end{array}}{384 \times 36283 \times 1289 \times 10^6}$$

= +30.1 mm

Final = -30.5 + 30.1 = -0.4 mm < span/250 = 26 mm

## Conclusions to EC2 Prestress

1. Only 1 value for tension class =  $f_{ctm}$
2. Zero tension if exposure > XC1
3. Prestress losses for initial relaxation and elastic shortening, plus shrinkage, creep and relaxation
4. Ultimate stress and strain equilibrium
5. Camber = immediate at transfer + creep
6. Deflections = static + creep